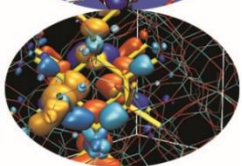
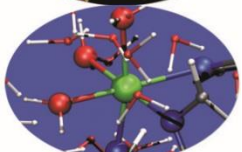
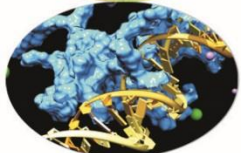
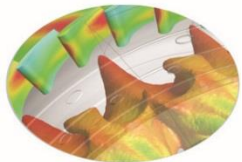


Simulation of detached flows using OpenFOAM®



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IEFLUIDS

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CETENA

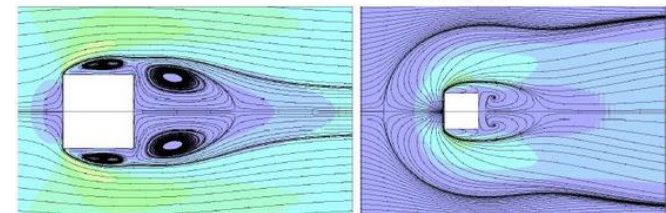
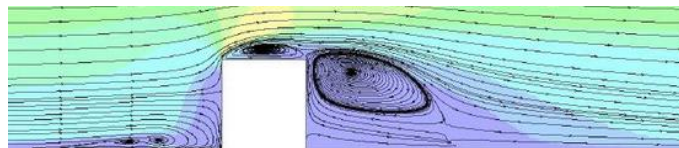
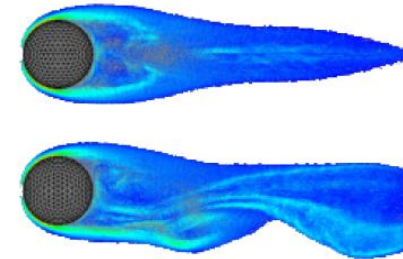
8 April 2016



- Flow separation is an important phenomena in industrial and environmental applications
- Most of the models having problem to predict the separation and reattachment points

Benchmark cases in order to test applicative meshes:

- ✓ Flow past a sphere
- ✓ Flow around Surface-Mounted cubical obstacle



Numerical scheme

Turbulence models:

□ RANS (standard $k - \varepsilon$ model)

$$\begin{cases} \frac{\partial}{\partial t}(\rho k) + \frac{\partial}{\partial x_i}(\rho k \bar{u}_i) = \frac{\partial}{\partial x_j} \left[\left(\mu + \frac{\mu_t}{\sigma_k} \right) \frac{\partial k}{\partial x_j} \right] + P_k - \rho \varepsilon \\ \frac{\partial}{\partial t}(\rho \varepsilon) + \frac{\partial}{\partial x_i}(\rho \varepsilon \bar{u}_i) = \frac{\partial}{\partial x_j} \left[\left(\mu + \frac{\mu_t}{\sigma_\varepsilon} \right) \frac{\partial \varepsilon}{\partial x_j} \right] + C_{1\varepsilon} \frac{\varepsilon}{k} P_k - C_{2\varepsilon}^* \rho \frac{\varepsilon^2}{k} \end{cases}$$

P_k is the rate of production of k by the mean flow

$$C_{2\varepsilon}^* = C_{2\varepsilon} + \frac{C_\mu \eta^3 (1 - \eta/\eta_0)}{1 + \beta \eta^3}$$

$\eta = Sk/\varepsilon$ with S the modulus of the mean rate-of-strain

$$\mu_t = \rho C_\mu \frac{k^2}{\varepsilon}$$

$$C_\mu = 0.09 \quad \sigma_k = 1.00 \quad \sigma_\varepsilon = 1.30 \quad C_{1\varepsilon} = 1.44 \quad C_{2\varepsilon} = 1.92$$



Numerical scheme

Turbulence models:

□ LES

- standard Smagorinsky : $\nu_{eff} = \nu + \nu_t$

$$\nu_T = (C_s \bar{\Delta})^2 |\bar{S}|, C_s = \sqrt{c_k \sqrt{2 \frac{c_k}{c_\epsilon}}}$$

- Smagorinsky with van Driest damping

$$\Delta = \min \left(\Delta_{mesh}, \left(\frac{k}{C_\Delta} \right) y (1 - e^{-y^+/A^+}) \right)$$

- dynamic Lagrangian

$$M_{ij} = 2\Delta^2 \left[\widehat{|\bar{S}| \bar{S}_{ij}} - 4|\bar{S}| \bar{S}_{ij} \right]$$

$$c_s^2(\mathbf{x}, t) = \frac{\mathcal{I}_{LM}}{\mathcal{I}_{MM}}$$

Meneveau et al (1996)

$$\mathcal{I}_{MM}(\mathbf{x}, t) = \int_{-\infty}^t M_{ij} M_{ij}(\mathbf{z}(t'), t') W(t - t') dt'$$

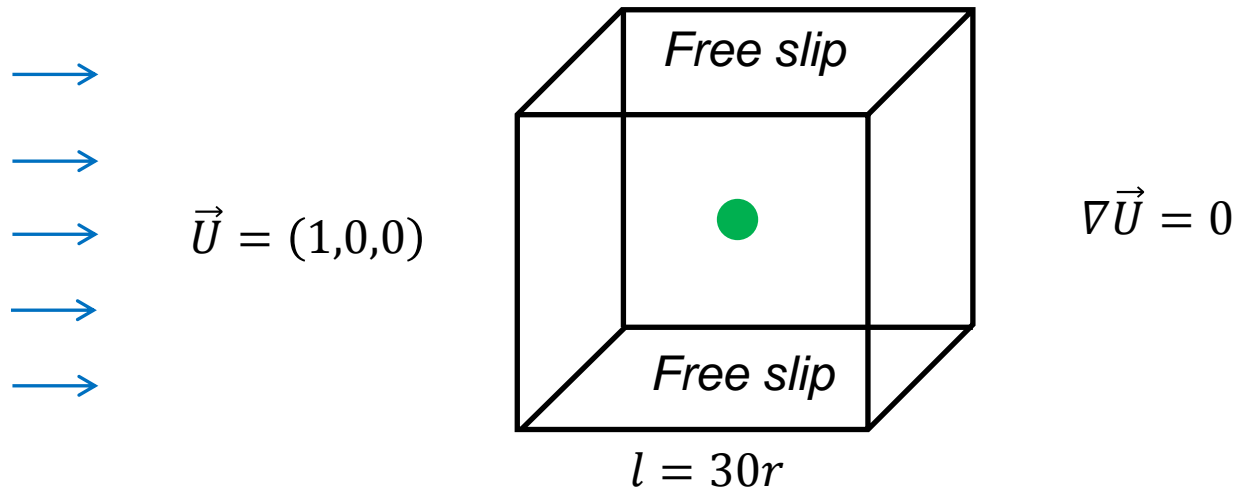
$$\mathcal{I}_{LM}(\mathbf{x}, t) = \int_{-\infty}^t L_{ij} M_{ij}(\mathbf{z}(t'), t') W(t - t') dt',$$

Numerical scheme

- Second order schemes
- Central difference for advective terms
- PISO algorithm for LES,
SIMPLE for RANS.

Problem description

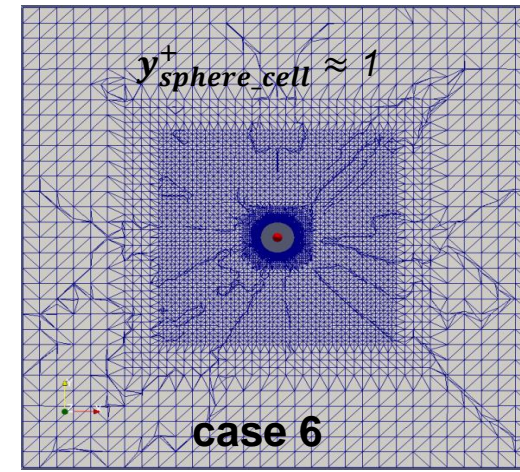
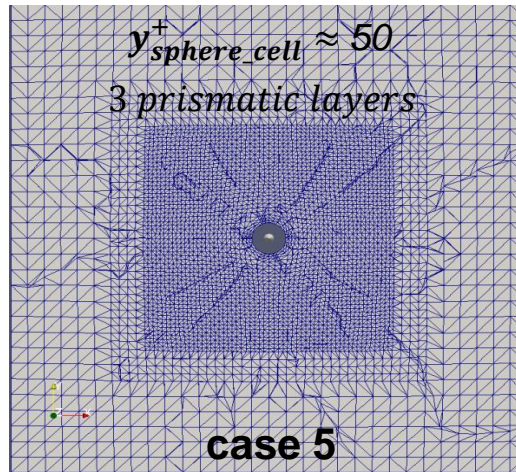
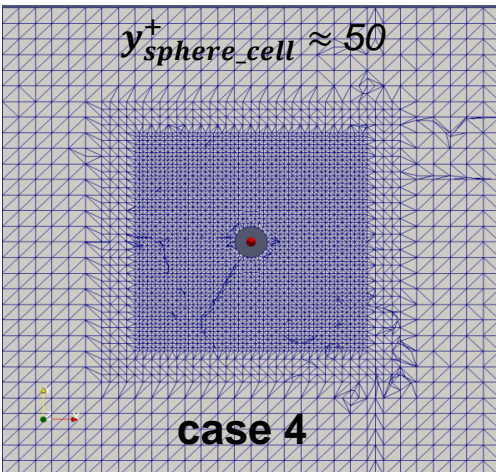
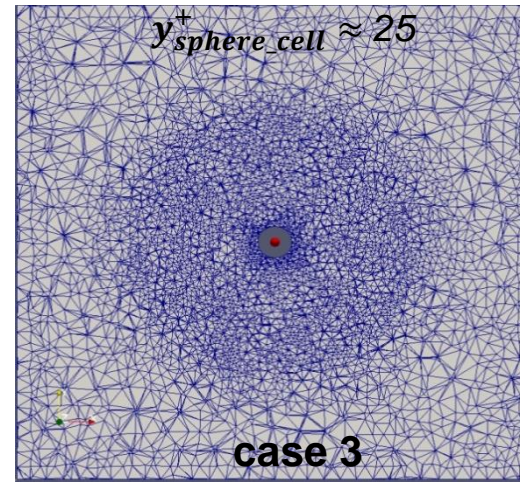
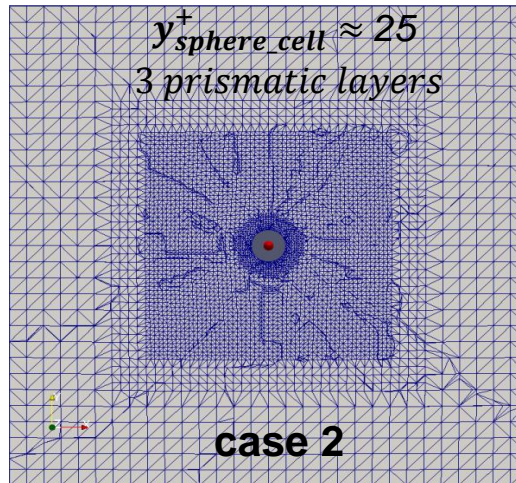
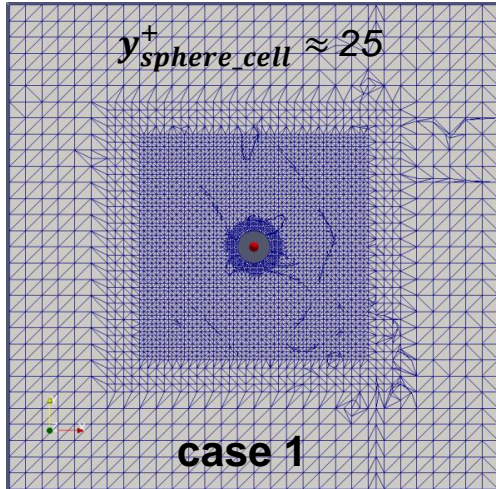
$$Re_D = 2rU/\nu = 10000, \quad r = 1 \text{ m.}$$



Constantinescu & Squires (2003)

Flow around sphere

30 × 30 × 30 cells



OpenFOAM

Grid:

- case 1**
- case 2**
- case 4**
- case 5**
- case 6**

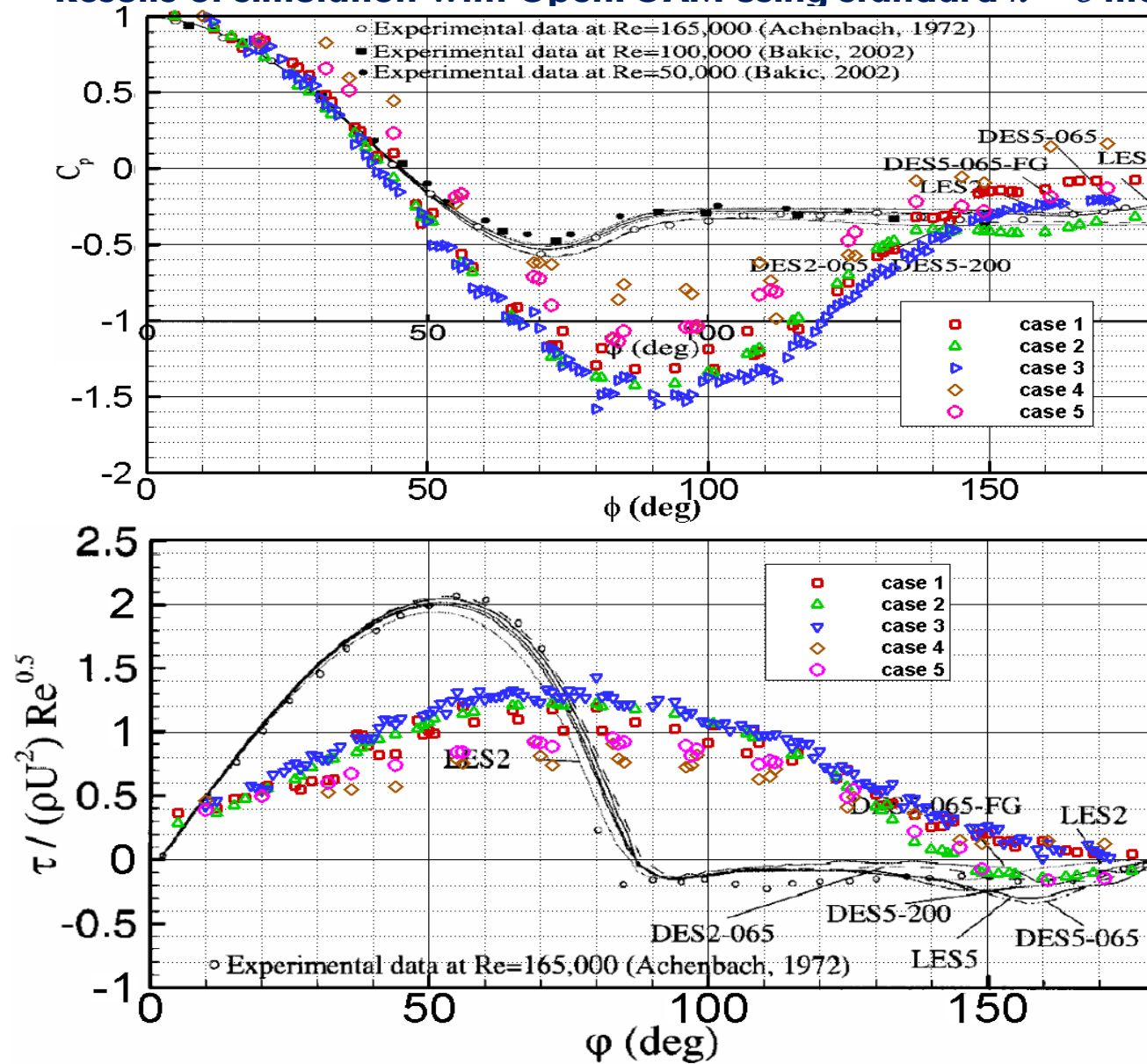
ANSYS ICEM

Grid:

- case 3**

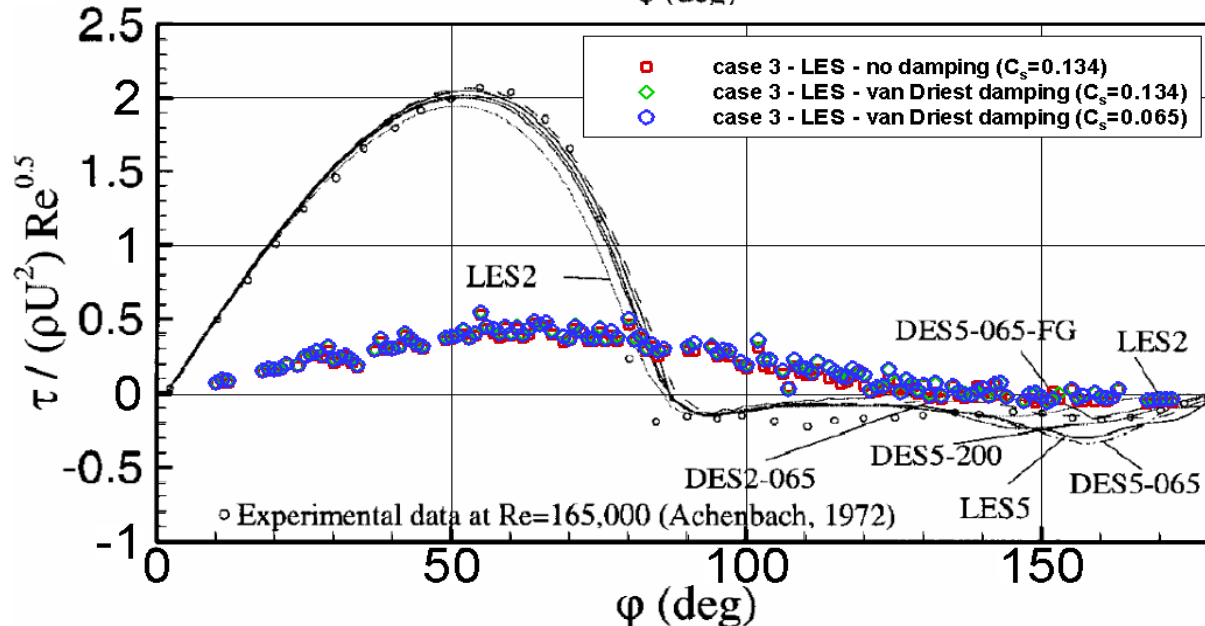
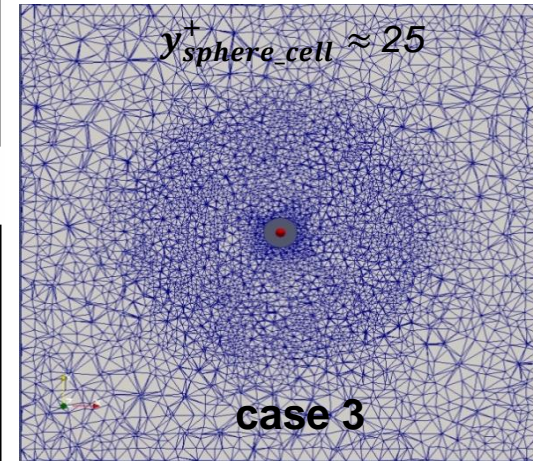
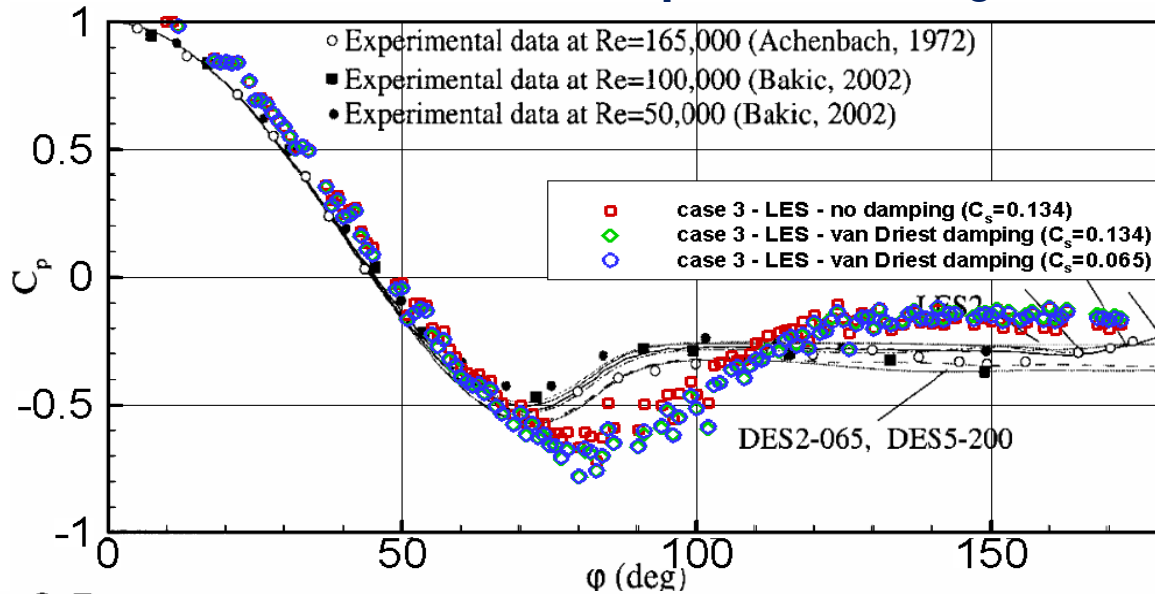
Flow around sphere

Results of simulation with OpenFOAM using standard $k - \varepsilon$ model



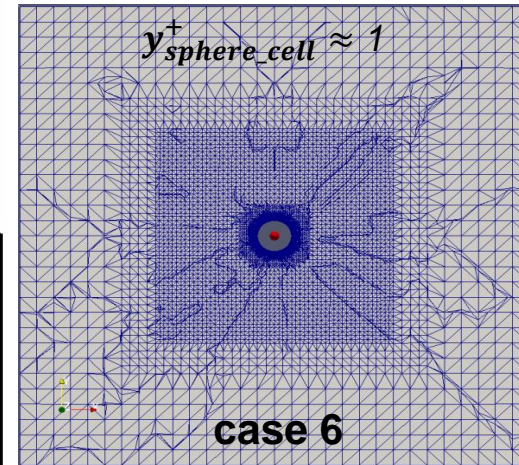
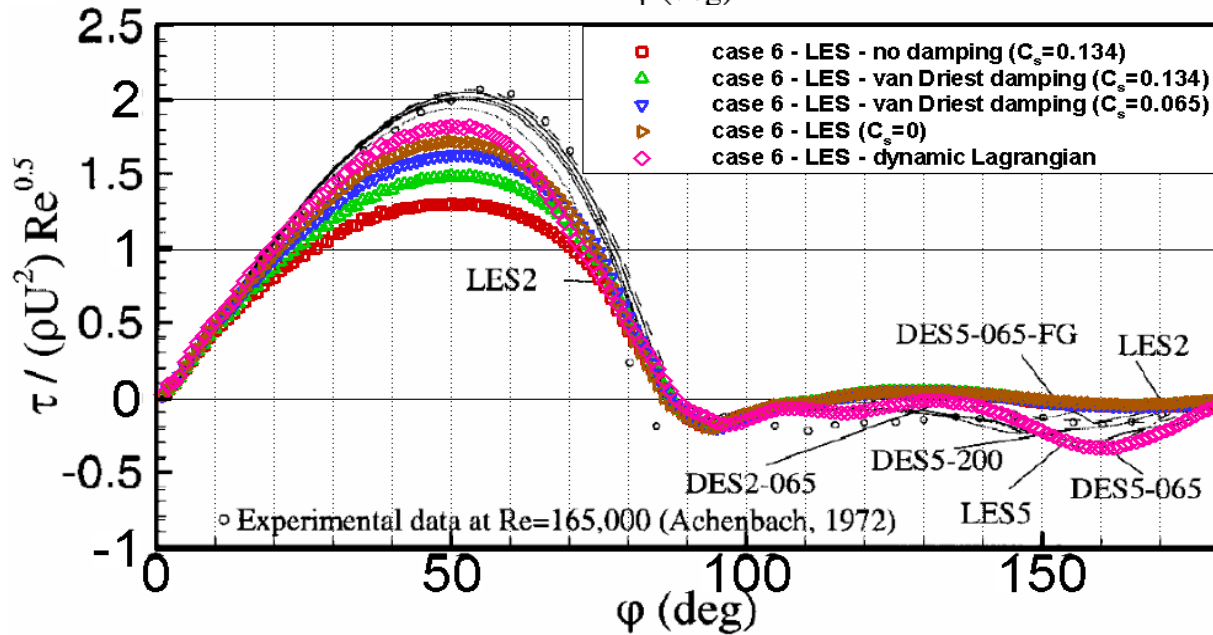
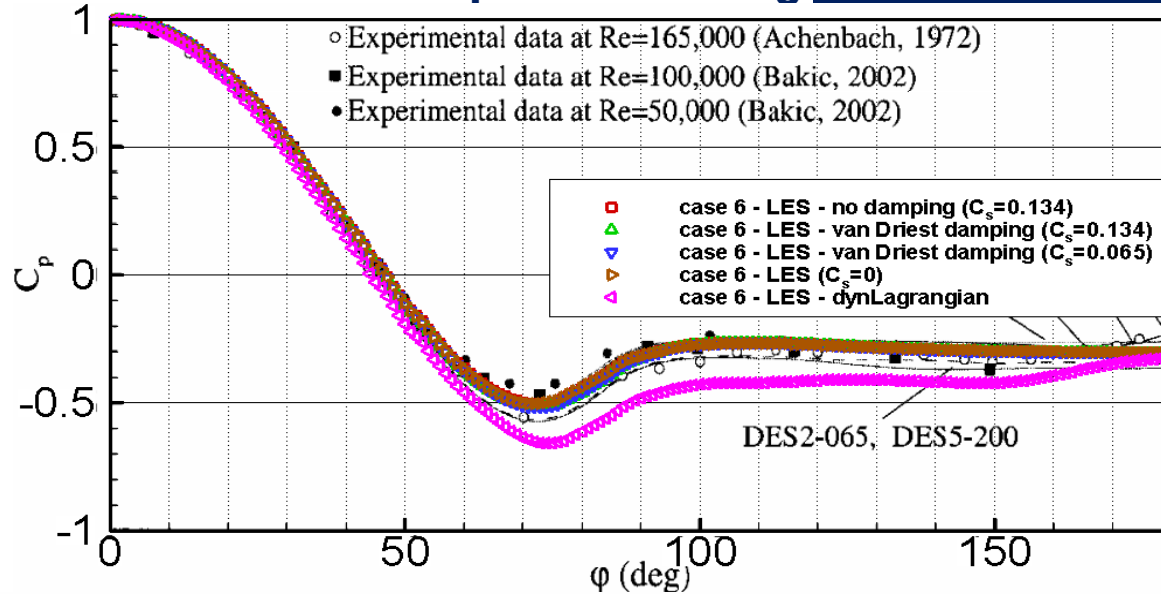
Flow around sphere

Results of simulation with OpenFOAM using LES for case 3



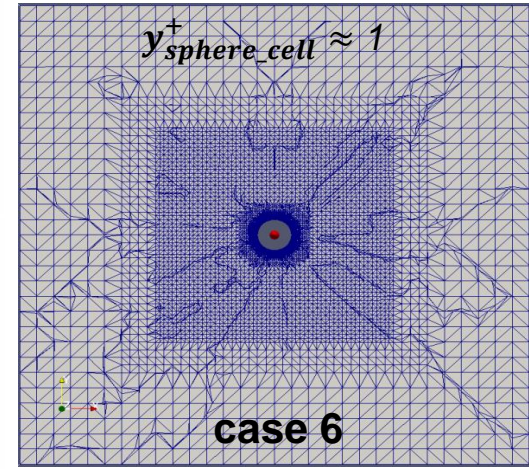
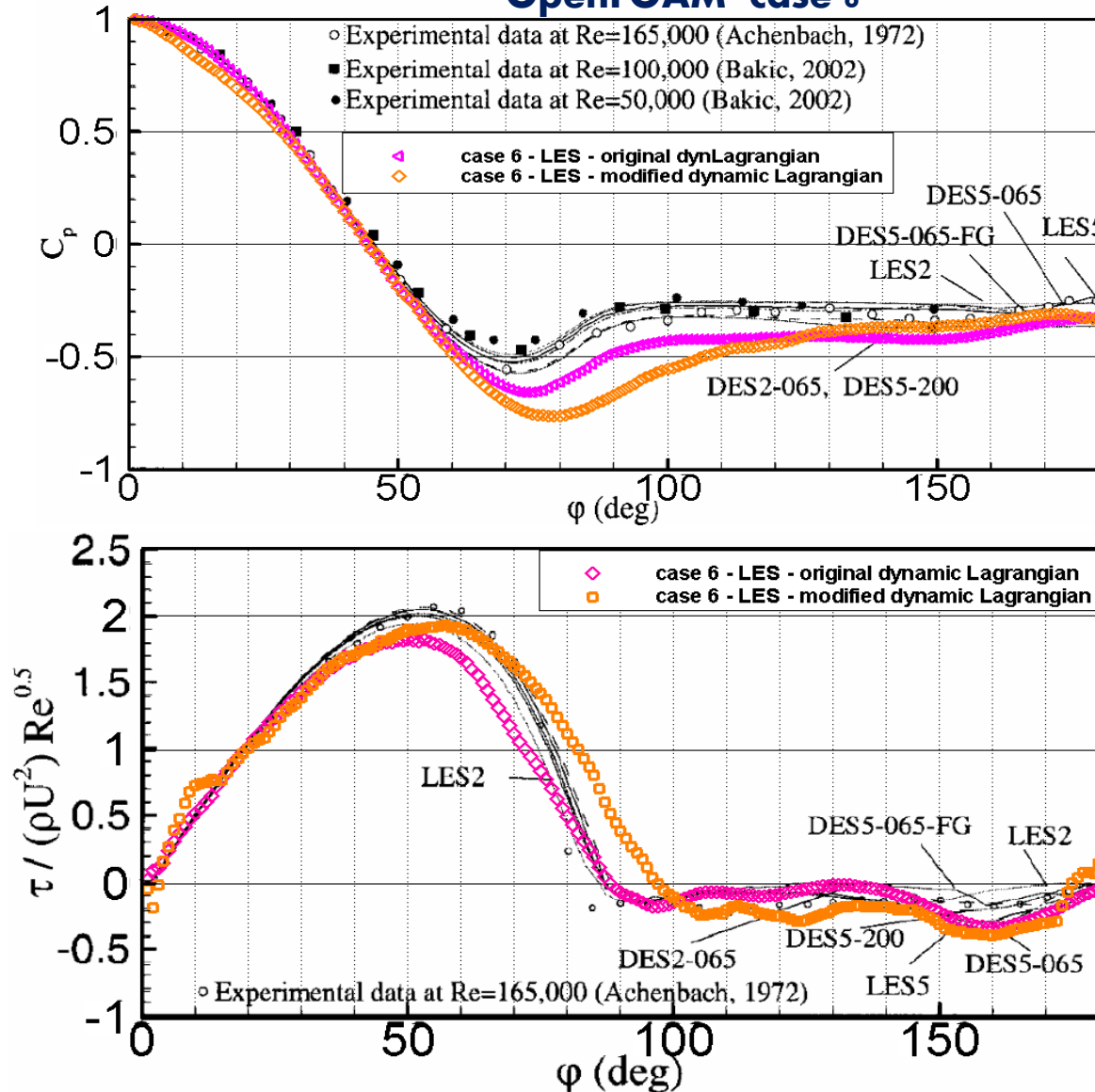
Flow around sphere

Result of simulation with OpenFOAM using *LES (Smagorinsky)* for case 6

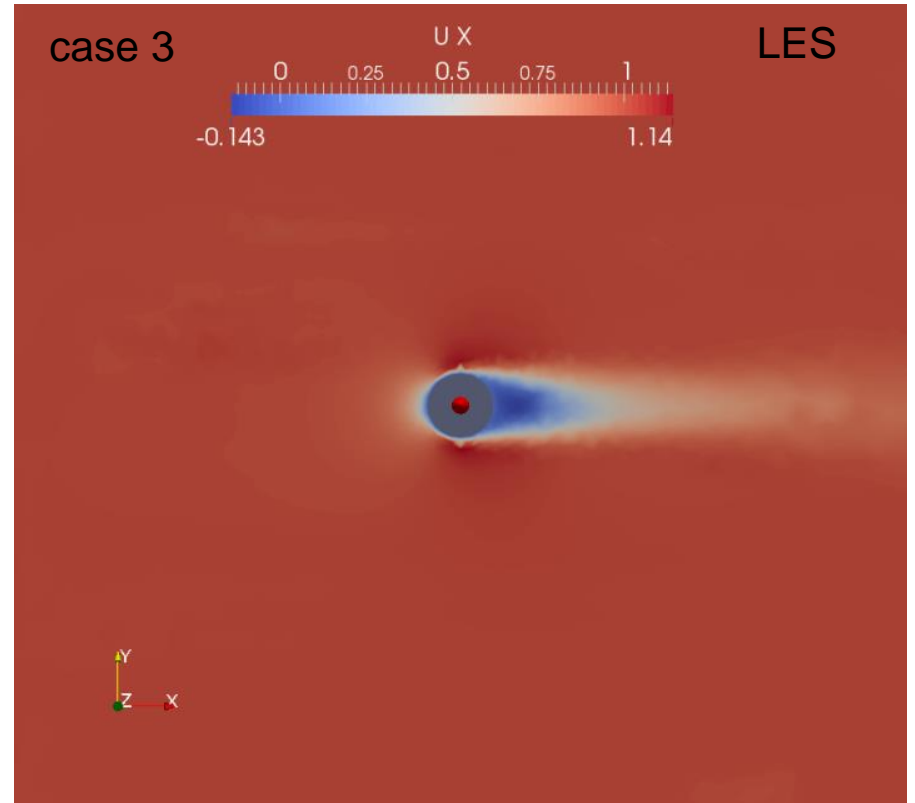
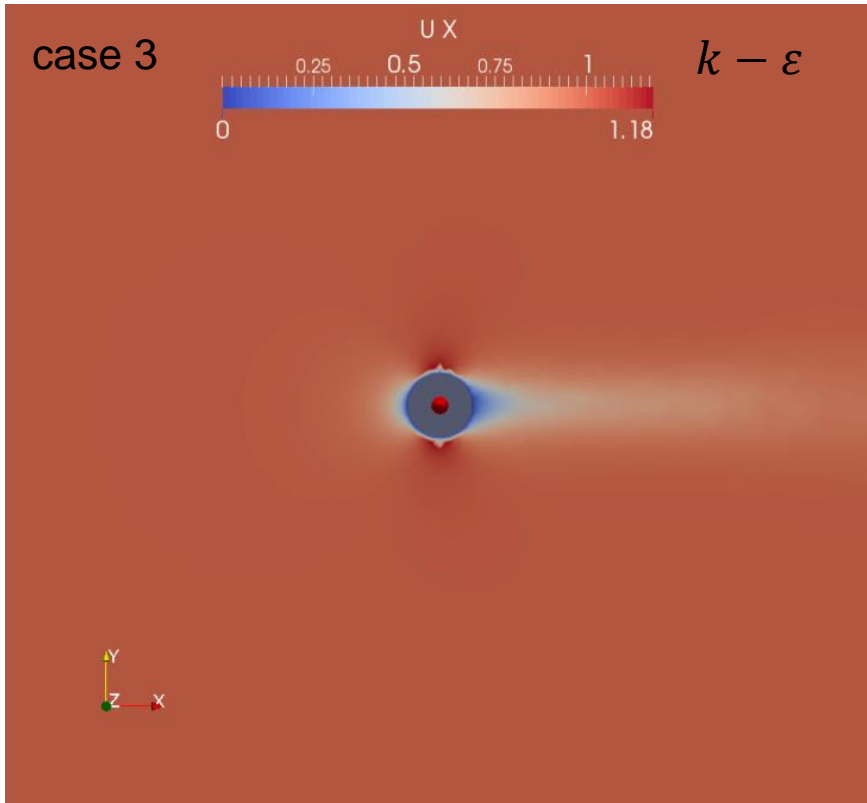


Flow around sphere

Results of simulation with LES-dynLagrangian : our modification and the original of **OpenFOAM case 6**



Flow around sphere

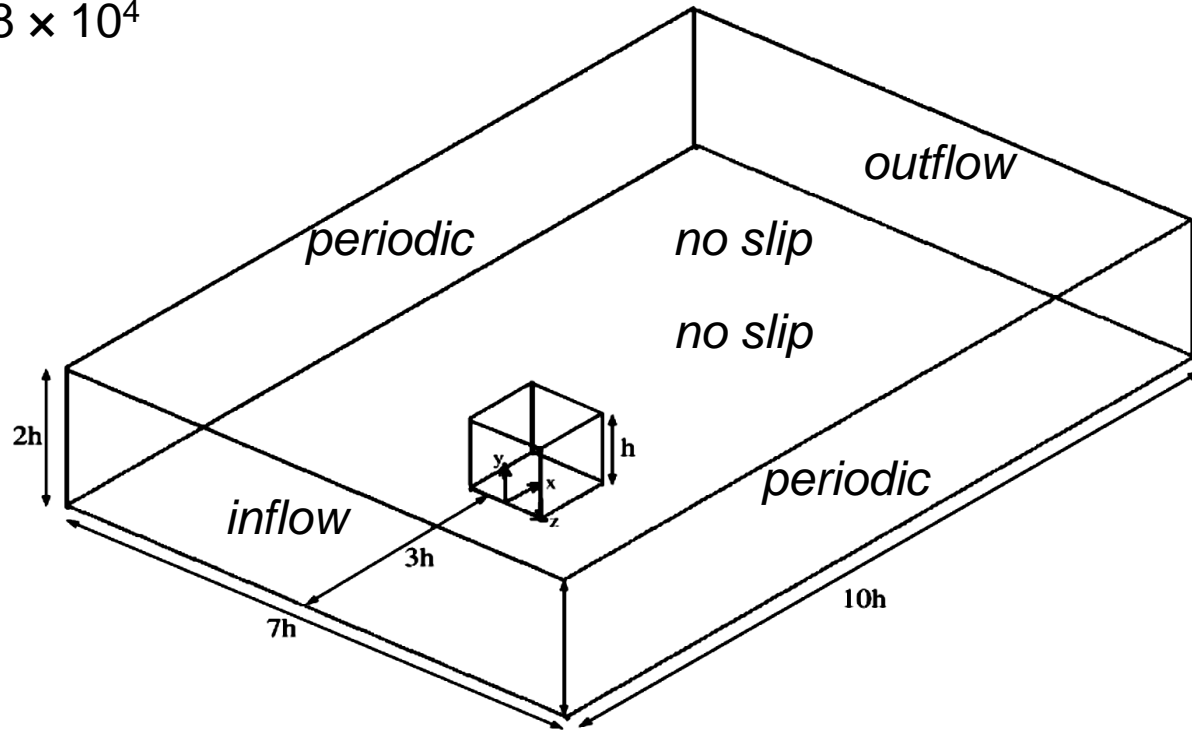


wall-mounted box

Problem description

$$Re_h = U_B h / \nu = 8 \times 10^4$$

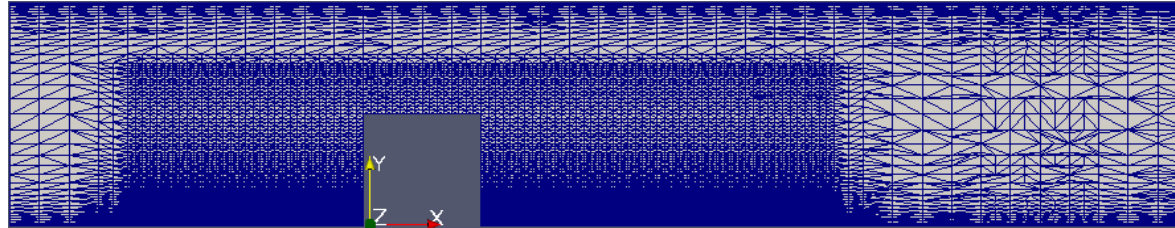
$$h = 0.025 \text{ m}$$



experiment by Martinuzzi and Tropea (1993)
 ERCOFTAC database

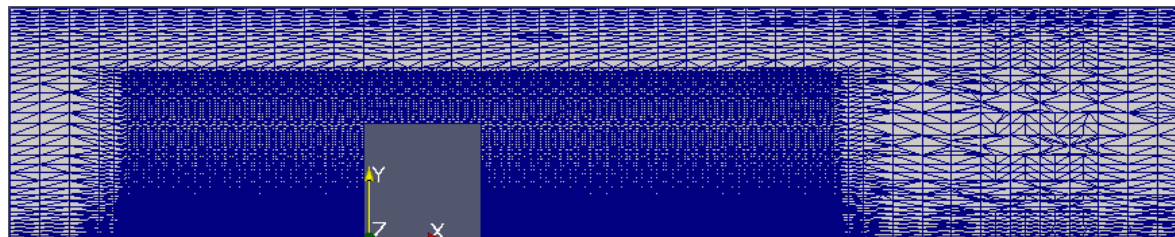
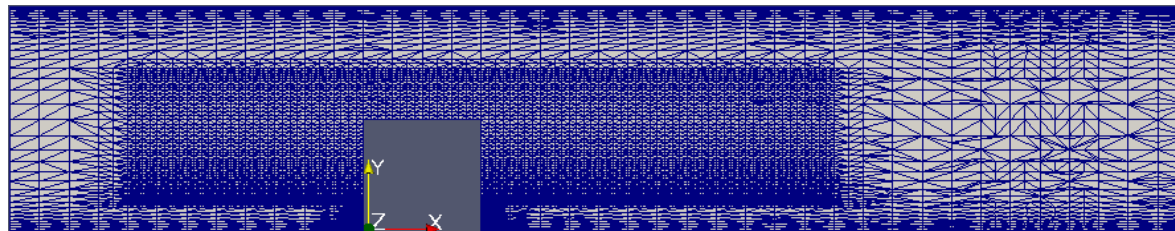
Flow over box

Grid generated by OpenFOAM

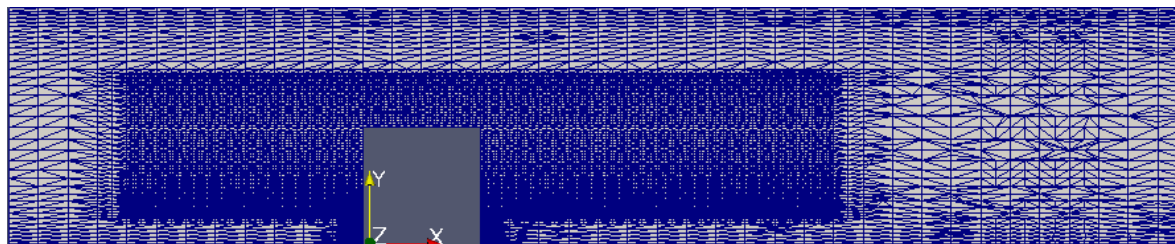


$y^+ \approx 64$ Near cube

Refinement box from bottom



Refinement box from bottom

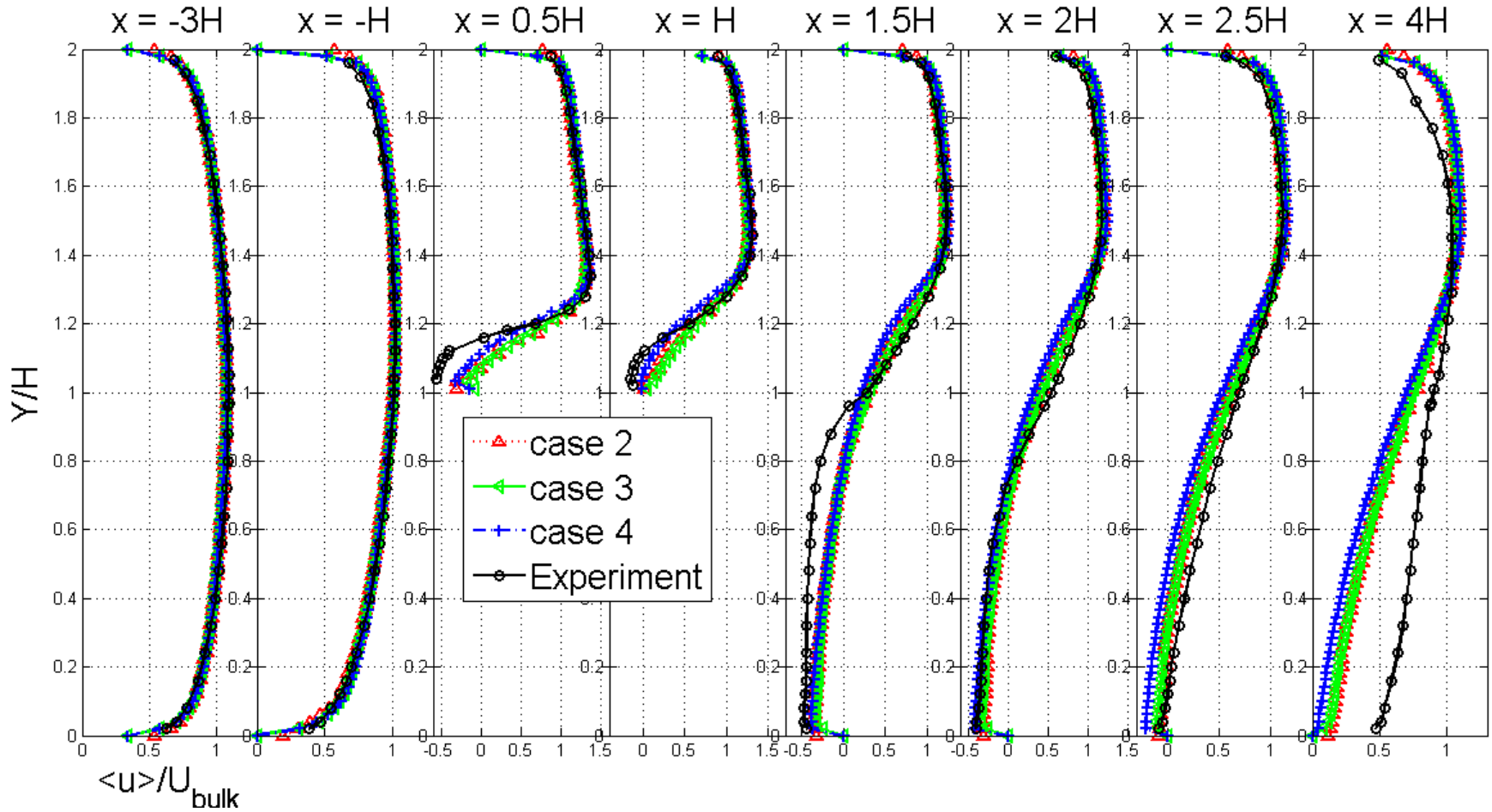


40 × 36 × 24 cells



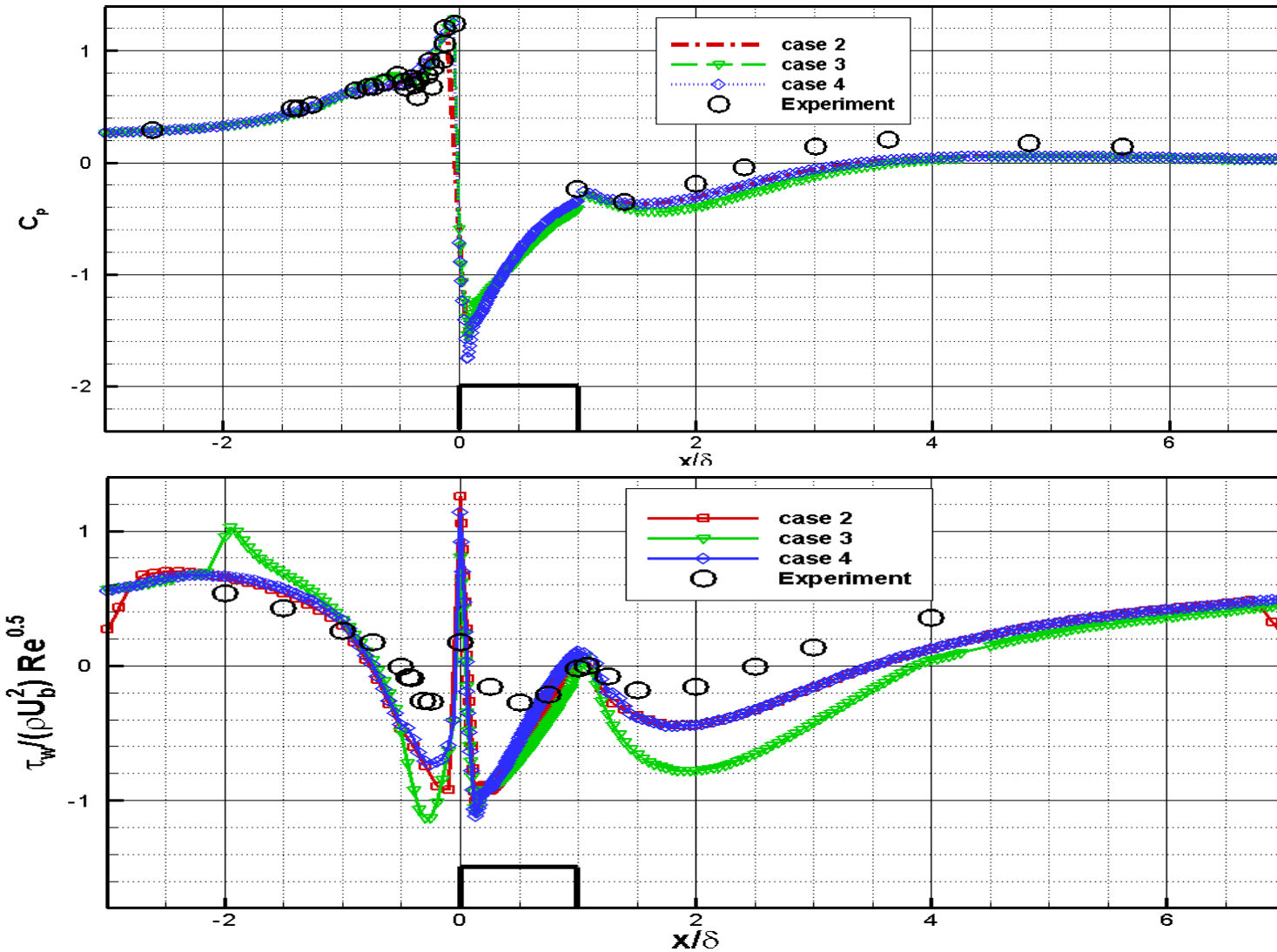
Flow over box

Results of simulation with OpenFOAM using standard $k - \varepsilon$ model



Flow over box

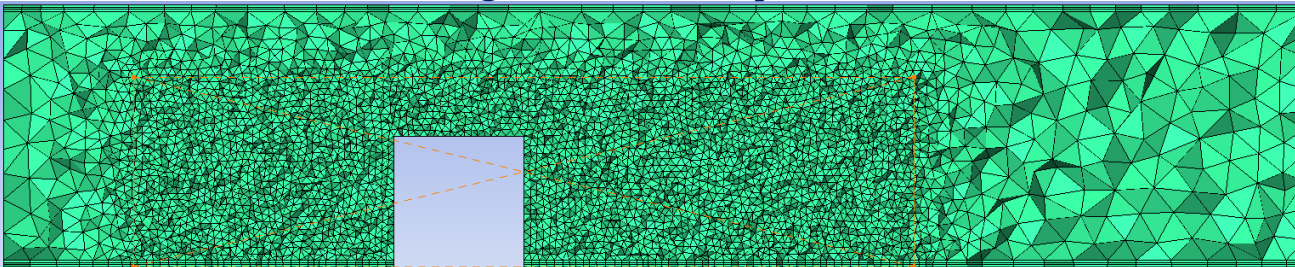
Results of simulation with OpenFOAM using standard $k - \varepsilon$ model



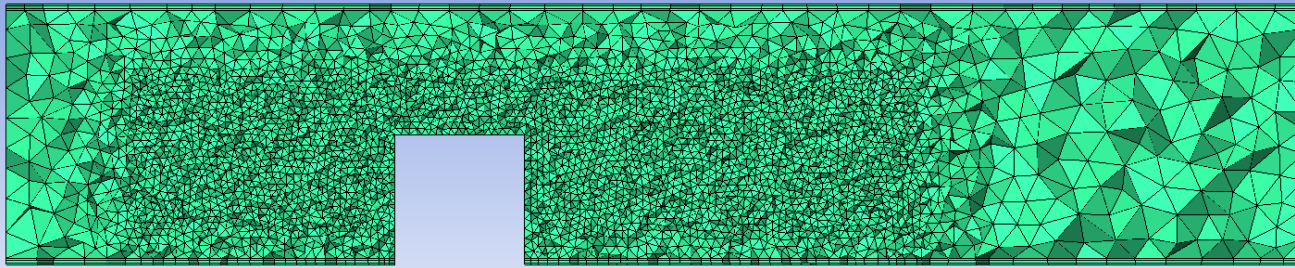
Flow over box

Grid generated by ANSYS ICEM

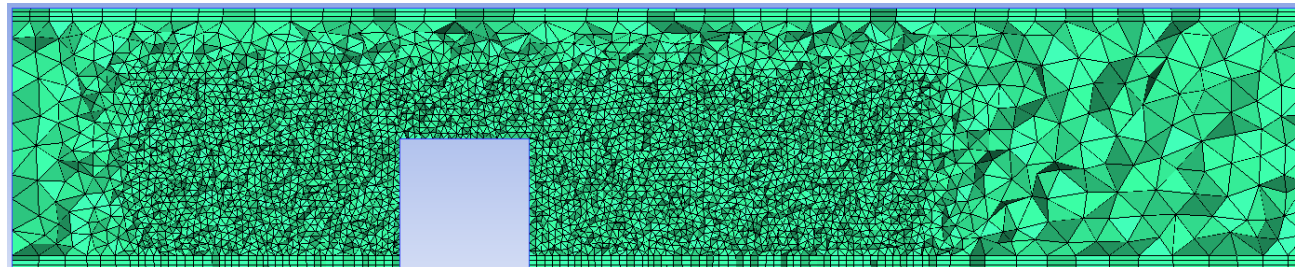
case 1



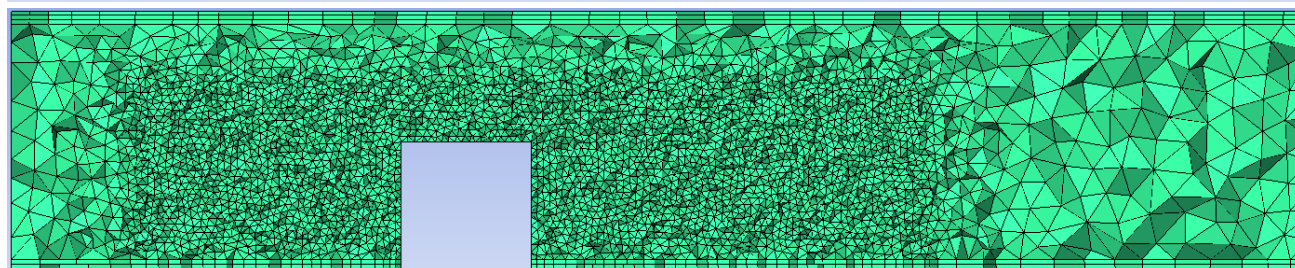
case 2



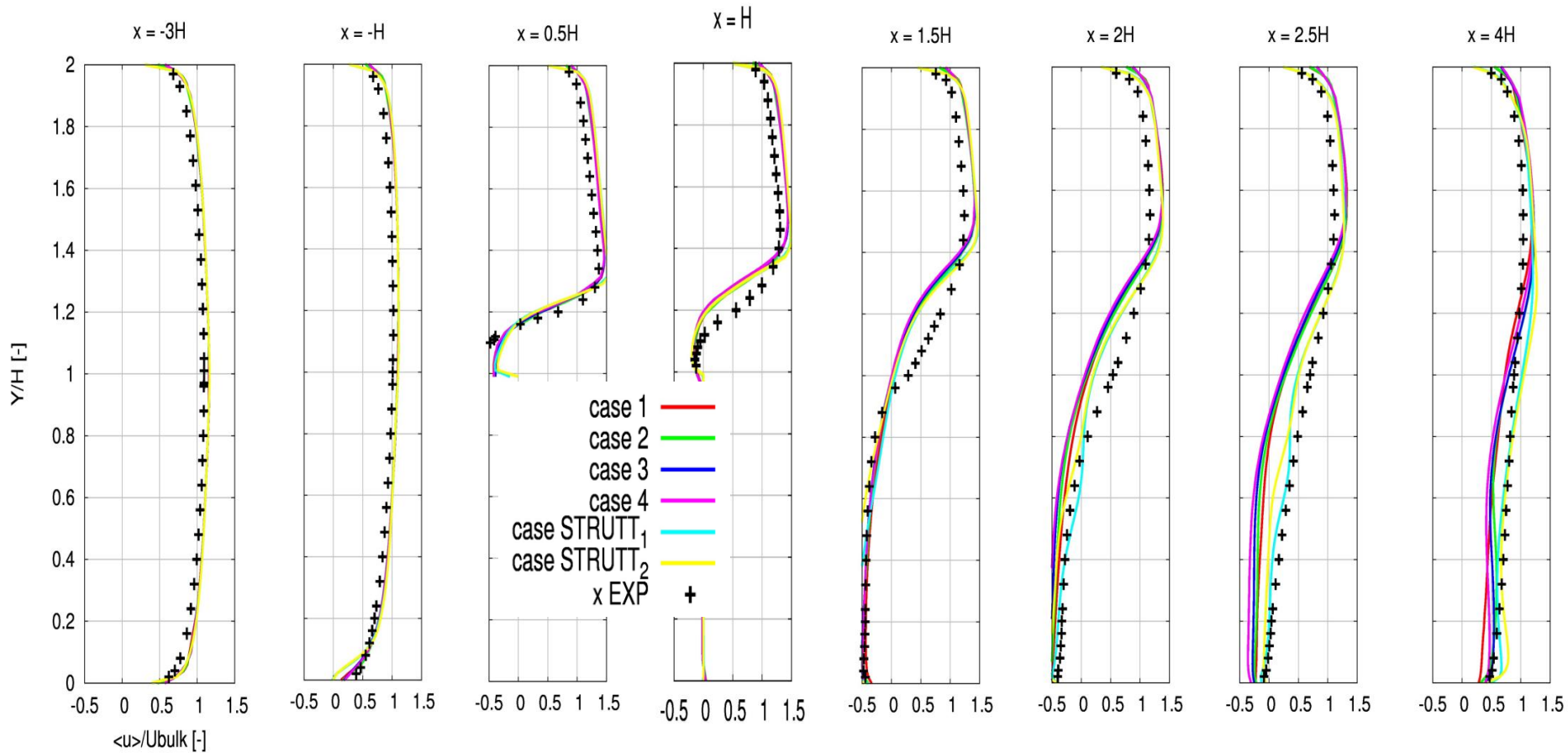
case 3



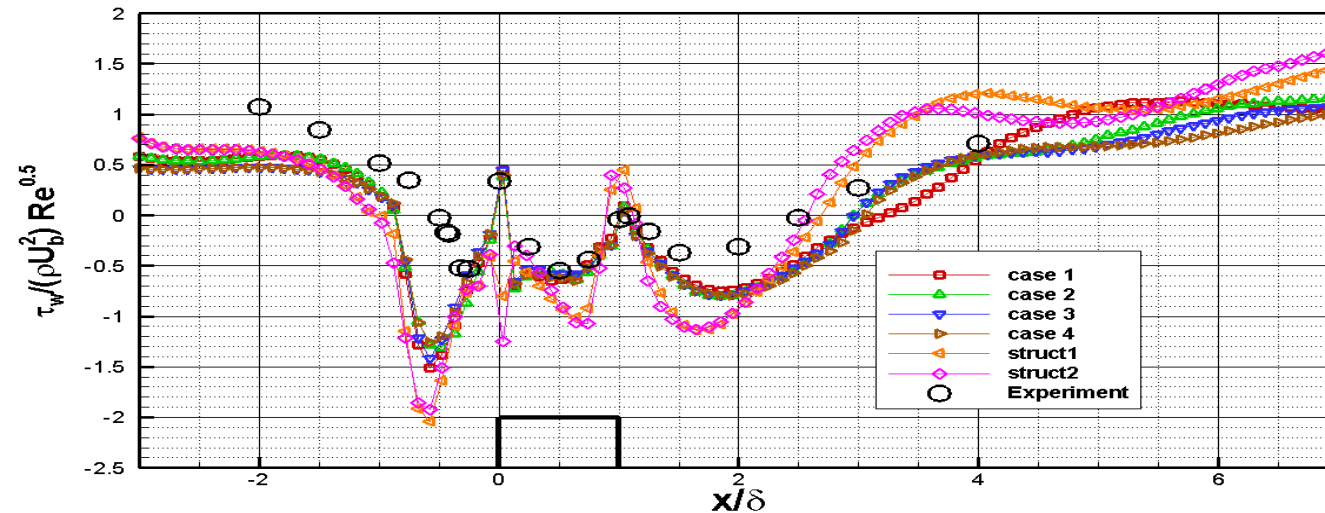
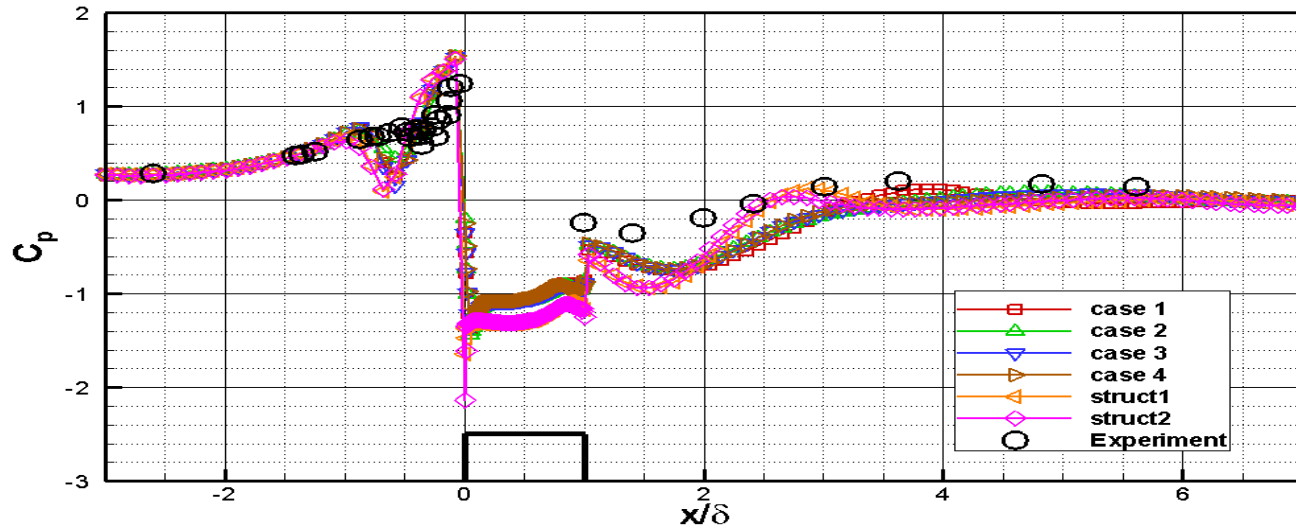
case 4



Result of simulation with ANSYS CFX using *SST* $k - \omega$ model



Results of simulation with ANSYSCFX using *SST* $k - \omega$ model



Flow over box

Wall-bounded fully developed turbulent flows passing around bluff-bodies.

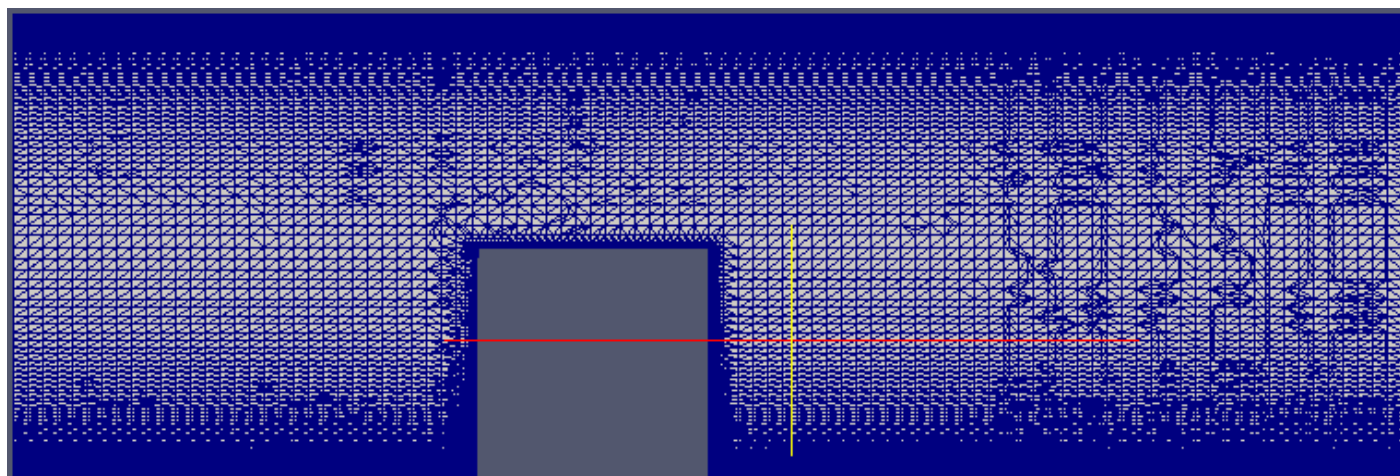
A random noise is added to the specific inlet velocity from a defined turbulence level

Inlet

```

{
Type turbulenceInlet;
referenceField nonuniform
fluctuationScale (0.02 0.01 0.01 )
Value nonuniform
}
  
```

$$u_p^+ = \begin{cases} \frac{1}{k} \log(y_p^+) + B, & y_p^+ > 11 \\ y_p^+, & y_p^+ \leq 11 \end{cases}$$

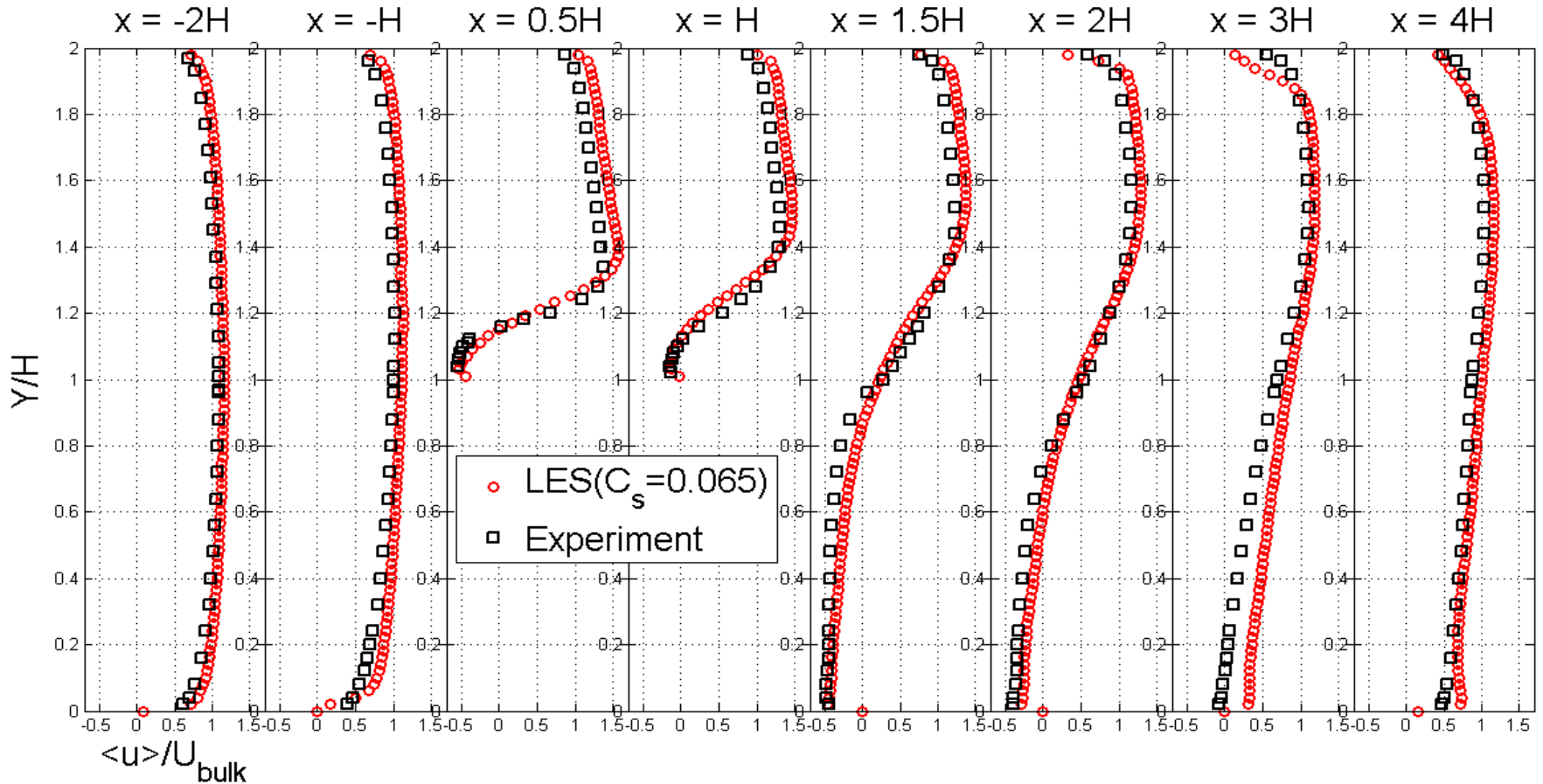


$6\delta \times 2\delta \times 7\delta$
94 × 120 × 160 cells

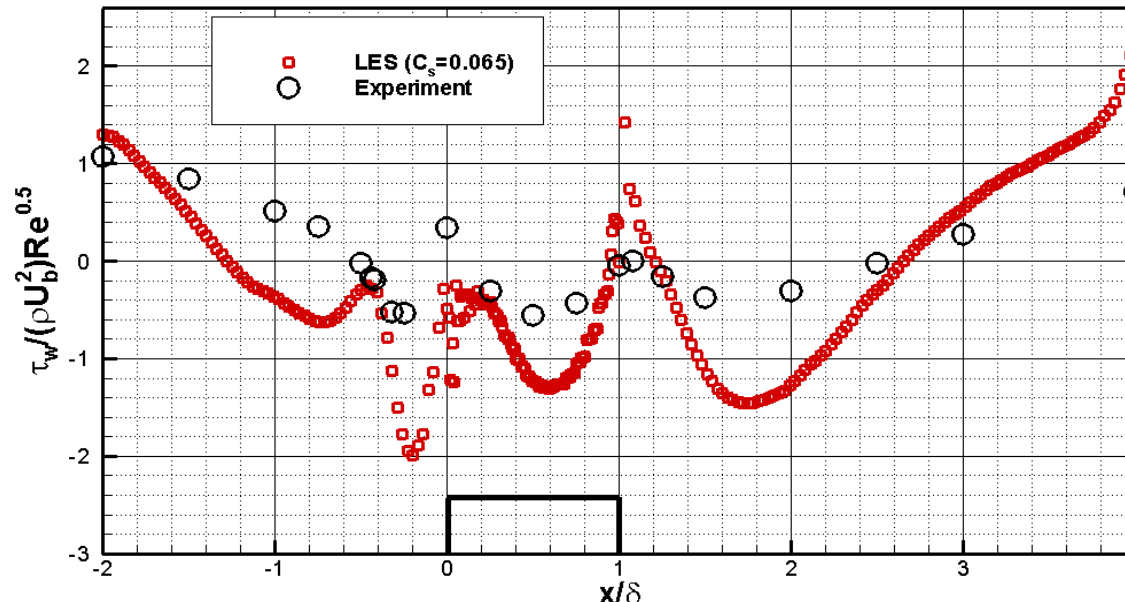
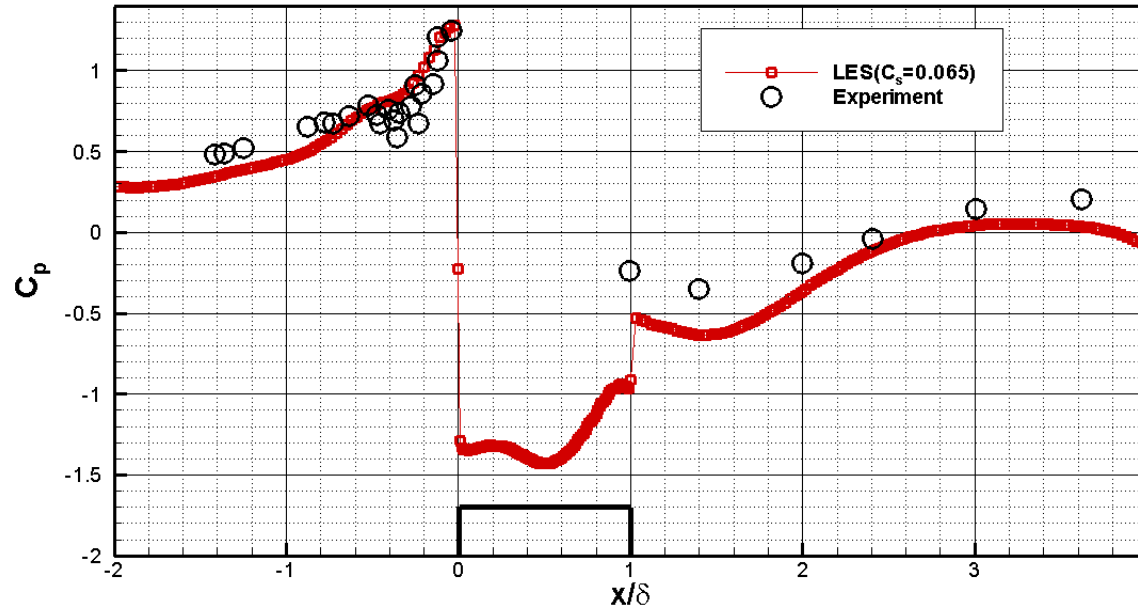


Flow over box

Simulation with OpenFOAM, using standard Smagorinsky (van Driest damping)

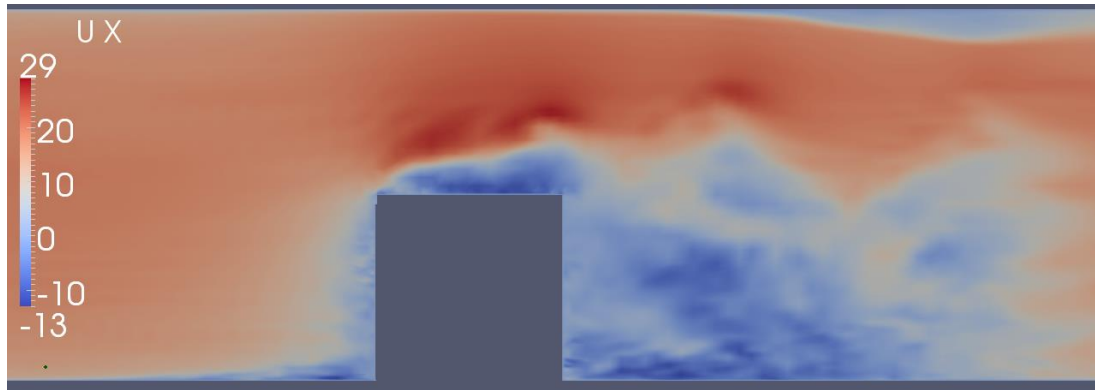


Flow over box

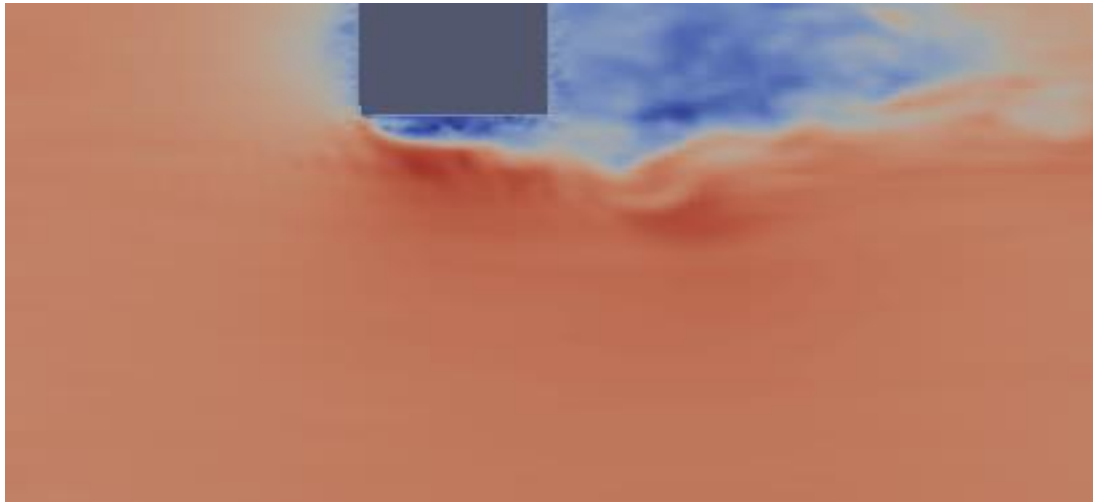


Flow over box

Instantaneous Flow field



Vertical cross section



Horizontal cross section

Why Wall layer model?

- **Wall bounded flows**, at very **high Reynolds** numbers, require **very fine grid** near wall to resolve viscous sub-layer
- The resolution even for **resolved LES** ($Re^{2.5}$) is comparable to **DNS** (Re^3), using wall function ($Re^{0.6}$) [Piomelli 2008]
- In applications where wall roughness is the rule rather than the exception, it does not make sense to describe the wall boundary in a deterministic sense.

Wall layer model

Filtered Navier Stokes equation for an incompressible flow:

$$\frac{\partial \bar{u}_i}{\partial t} + \frac{\partial (\bar{u}_i \bar{u}_j)}{\partial x_j} = -\frac{1}{\rho} \frac{\partial \bar{p}}{\partial x_i} + \frac{\partial}{\partial x_j} \left((\nu + \nu_T) \frac{\partial \bar{u}_i}{\partial x_j} \right)$$

Equilibrium stress model:

$$u_p^+ = \begin{cases} \frac{1}{k} \log(y_p^+) + B, & y_p^+ > 11 \\ y_p^+, & y_p^+ \leq 11 \end{cases}$$

$$u_p^+ = u_p / \mathbf{u}_\tau = \sqrt{(u^2 + w^2)} / \mathbf{u}_\tau, \quad k = 0.41, \quad B = 5.1$$

$$\text{Wall shear stress: } \tau_w = \rho u_\tau^2 \rightarrow \begin{cases} \tau_{w,x} = \rho u_\tau^2 \frac{u}{u_p} \\ \tau_{w,z} = \rho u_\tau^2 \frac{w}{u_p} \end{cases}$$

In x direction:

$$(\nu + \nu_T) \frac{\partial \bar{u}_1}{\partial x_2} = \tau_{w,x}$$

In z direction:

$$(\nu + \nu_T) \frac{\partial \bar{u}_3}{\partial x_2} = \tau_{w,z}$$

Why Wall layer model?

P in the log region:

$$\bar{S}_{12} = \frac{u_\tau}{k y_p} \frac{\bar{u}}{u_p} + \frac{\partial \bar{v}}{\partial x}, \quad \bar{S}_{32} = \frac{u_\tau}{k y_p} \frac{\bar{w}}{u_p} + \frac{\partial \bar{v}}{\partial z}$$

P in the viscous layer:

$$\bar{S}_{12} = \frac{u_\tau^2}{\nu} \frac{\bar{u}}{u_p} + \frac{\partial \bar{v}}{\partial x}, \quad \bar{S}_{32} = \frac{u_\tau^2}{\nu} \frac{\bar{w}}{u_p} + \frac{\partial \bar{v}}{\partial z}$$

$$\mathbf{v}_T = (C_s \bar{\Delta})^2 |\bar{\mathbf{S}}|$$

- ❑ **WMLES for flows with separation**
- ❑ **Proper Turbulent inflow for LES**
- ❑ **Applying LES (dynamic Lagrangian) to simulate flow around wall-mounted cube**
- ❑ **Applying DES**

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