

# PHYSICS AND HIGH PERFORMANCE COMPUTATION OF TURBULENT FLOWS WITH INTERFACES

FRANCESCO ZONTA

Vienna University of Technology  
&  
University of Udine, Udine

[francesco.zonta@tuwien.ac.at](mailto:francesco.zonta@tuwien.ac.at)

## ACKNOWLEDGMENTS

THANKS TO:

PROF. ALFREDO SOLDATI,

PROF. MIGUEL ONORATO,

DR. MARCO DE PAOLI,

DR. ALESSIO ROCCON

**\*SPECIAL THANKS TO CINECA**

FOR GENEROUS ALLOWANCE OF COMPUTER RESOURCES

(ISCRA & PRACE PROJECTS)

## OUTLINE OF THE SEMINAR

- PART I            INTRODUCTION: SPECTRAL AND PSEUDO-SPECTRAL METHODS
  - PART II          THERMALLY STRATIFIED FLUIDS: INTERNAL WAVES AND BEYOND
  - PART III         SURFACE WAVES
  - PART IV          DROPLETS IN TURBULENCE
  - PART V           CARBON CAPTURE
- DISCUSSION ON NUMERICS & PHYSICS INVOLVED
- (ALWAYS WITH **AN EYE ON APPLICATIONS**)

## SPECTRAL AND PSEUDO-SPECTRAL

METHODS TO DISCRETIZE THE DIFFERENTIAL OPERATORS

IDEA: APPROXIMATE A FUNCTION (UNKNOWN, WHICH SATISFIES PDE+BC)

USING A LINEAR COMBINATION OF "TEST" FUNCTIONS THAT ARE GLOBAL

APPROXIMATION

$$u(x) = \sum_{k=0}^N c_k \varphi_k(x)$$

Common also to finite difference  
and finite element methods

For spectral methods:  
Global functions (defined in each node,  
and not equal to zero!)

PUTTING INTO THE PDE EQUATION+BC:

$$Lu(x) = s(x) \quad \xleftarrow{\text{DOMAIN}}$$

$$Bu(y) = 0 \quad \xleftarrow{\text{BOUNDARY}}$$

SOLUTION  $\bar{u}$

$L\bar{u} - s$  MINIMIZES THE RESIDUAL

WE WILL OBTAIN THE SOLUTION ( $c_k$ )

## SPECTRAL METHODS AT A GLANCE:

- Assume a basis (test functions; depend on the problem)
- Assume a number N of polynomials such that

$$u(x) \approx \tilde{u}(x) = \sum_{k=0}^N c_k \varphi_k(x)$$

- Substitute it into the pde+bc
- Minimize the residual to find the coefficients of the spectral representation

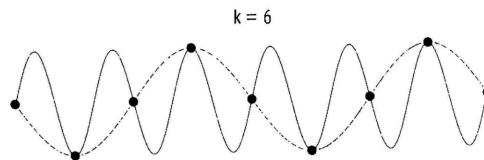
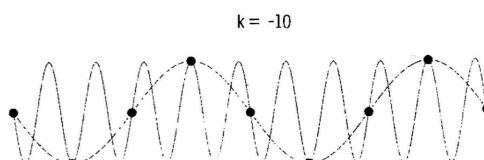
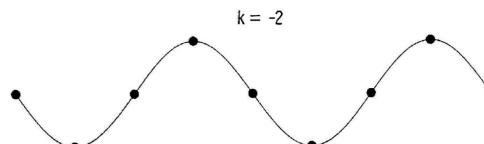
Tau	Minimize the residual in each node
Galerkin	Minimize the integral of the residual (over the domain)

BC for tau methods: add BC to the system (extra equations)

BC Galerkin: combine the basic functions to form new functions fulfilling BC

## WHY PSEUDO-SPECTRAL?

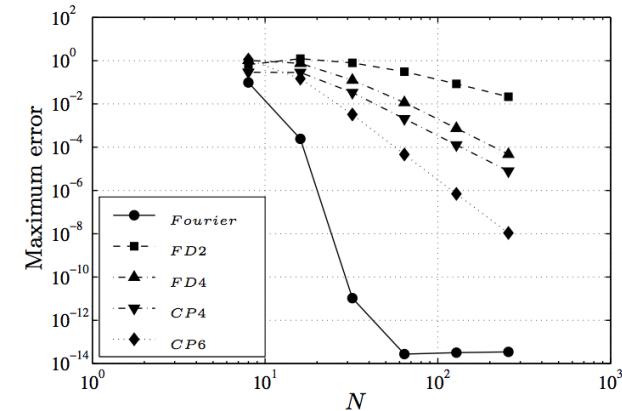
- Products in modal space: convolution  $O(N^2)$
- Transform into physical space, multiply and back to modal  $O(N \log_2 N)$
- Aliasing error (2/3 rules)

Aliasing of  $\sin(-2x)$  wave by  $\sin(6x)$  waveAliasing of  $\sin(-2x)$  wave by  $\sin(10x)$  wave

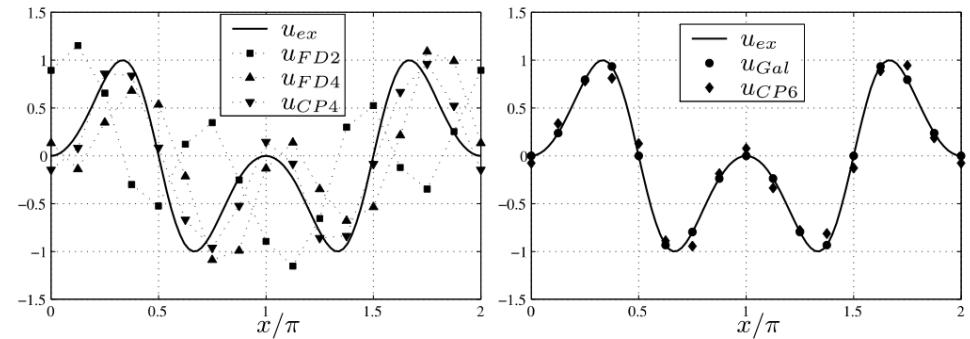
Canuto et al (1998)

## PRO AND CON

- In general, spectral accuracy and convergence
- With FFTW, good performances



- Aliasing
- Not easy to code
- Usually less flexible (simple geometry). But.....



Canuto et al (2010)

$$\frac{\partial u_i}{\partial x_i} = 0$$

$$\frac{\partial u_i}{\partial t} = S_i + \frac{1}{\text{Re}_\tau} \frac{\partial^2 u_i}{\partial x_j^2} - \frac{\partial p}{\partial x_i}$$

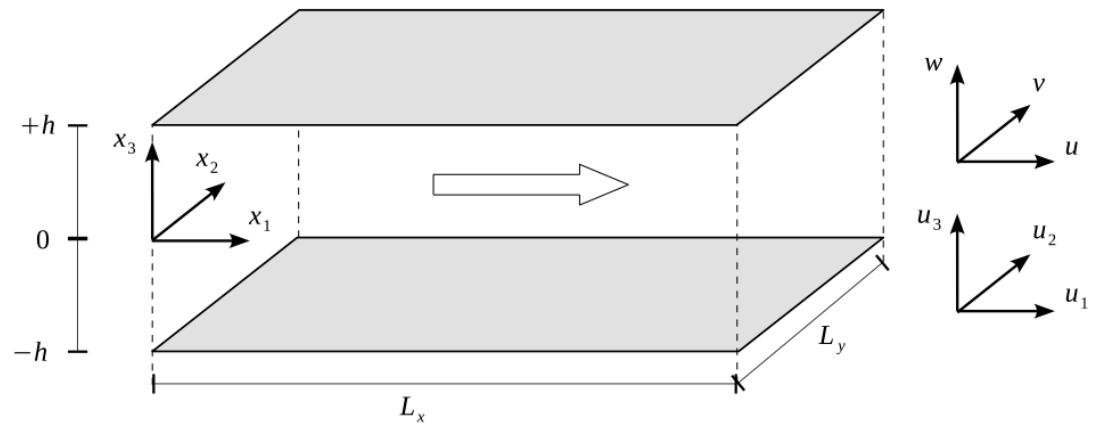
$$S_i = -\frac{\partial u_1 u_j}{\partial x_j} + \delta_{1,j}$$

Curl

$$\frac{\partial \omega}{\partial t} = \nabla \times S + \frac{1}{\text{Re}_\tau} \nabla^2 \omega$$

2<sup>nd</sup> Curl+Continuity+Vectorial Identity

$$\nabla \times (\nabla \times u) = \nabla (\nabla \cdot u) - \nabla^2 u$$



$$\frac{\partial (\nabla^2 u)}{\partial t} = \nabla^2 S - \nabla (\nabla \cdot S) + \frac{1}{\text{Re}_\tau} \nabla^4 u$$

This leads to the following system:

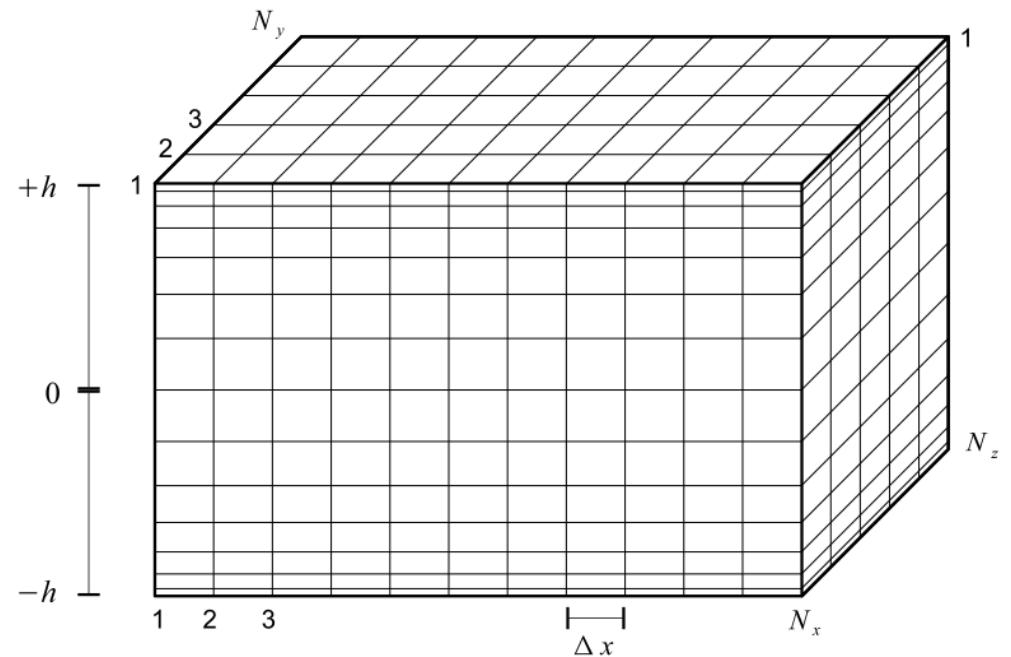
$$\frac{\partial \omega_3}{\partial t} = \frac{\partial S_2}{\partial x_1} - \frac{\partial S_1}{\partial x_2} + \frac{1}{\text{Re}_\tau} \nabla^2 \omega_3$$
$$\frac{\partial (\nabla^2 u_3)}{\partial t} = \nabla^2 S_3 - \frac{\partial}{\partial x_3} \left( \frac{\partial S_j}{\partial x_j} \right) + \frac{1}{\text{Re}_\tau} \nabla^4 u_3$$
$$\frac{\partial u_1}{\partial x_1} + \frac{\partial u_2}{\partial x_2} = - \frac{\partial u_3}{\partial x_3}$$
$$\frac{\partial u_2}{\partial x_1} - \frac{\partial u_1}{\partial x_2} = \omega_3$$

## Discretization strategy

$$x(i) = (i - 1) \frac{L_x}{N_x - 1} \rightarrow i = 1, \dots, N_x$$

$$y(j) = (j - 1) \frac{L_y}{N_y - 1} \rightarrow j = 1, \dots, N_y$$

$$z(k) = \cos\left(\frac{k - 1}{N_z - 1} \pi\right) \rightarrow k = 1, \dots, N_z$$

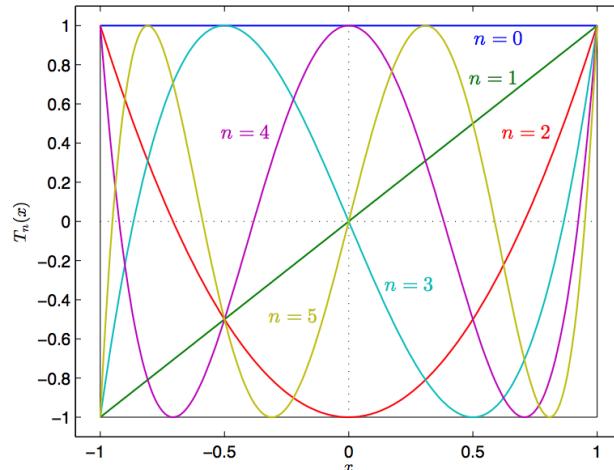


Spatial discretization: Fourier +

$$f(x_1, x_2, x_3) = \sum_{n_1} \sum_{n_2} \hat{f}(k_1, k_2, x_3) e^{i(k_1 x_1 + k_2 x_2)} \quad k_1 = \frac{2\pi n_1}{L_x}; k_2 = \frac{2\pi n_2}{L_y}$$

Chebyshev

$$f(x_1, x_2, x_3) = \sum_{n_1} \sum_{n_2} \sum_{n_3} \hat{f}(k_1, k_2, n_3) T_{n_3}(x_3) e^{i(k_1 x_1 + k_2 x_2)}$$



$$T_{n_3}(x_3) = \cos[n_3 \cos^{-1}(x_3/h)]$$

We get the following discrete equations:

$$ik_1\hat{u}_1 + ik_2\hat{u}_2 + \frac{\partial}{\partial x_3}\hat{u}_3 = 0 \quad \text{Continuity and vorticity}$$

$$\hat{\omega}_3 = ik_1\hat{u}_2 - ik_2\hat{u}_1$$

$$\frac{\partial \hat{\omega}_3}{\partial t} = ik_1\hat{S}_2 - ik_2\hat{S}_1 + \frac{1}{\text{Re}_\tau} \left( \frac{\partial^2}{\partial x_3^2} - k^2 \right) \hat{\omega}_3 \quad \text{Vorticity transport}$$

$$\boxed{\frac{\partial \hat{\omega}_3}{\partial t} = \xi + \psi}$$

Splitting of the operator  
Convective terms: Adams Bashfort  
Diffusive terms: cranck-Nicolson

$$\hat{\omega}_3^{n+1} = \hat{\omega}_3^n + \Delta t \left( \frac{3}{2}\xi^n - \frac{1}{2}\xi^{n-1} \right) + \frac{\Delta t}{2} (\psi^{n+1} + \psi^n)$$

After some algebra

$$\begin{aligned}\frac{\hat{\omega}_3^{n+1} - \hat{\omega}_3^n}{\Delta t} &= ik_1 \left[ \left( \frac{3}{2} \hat{S}_2^n - \frac{1}{2} \hat{S}_2^{n-1} \right) + \frac{1}{2 \operatorname{Re}_\tau} \left( \frac{\partial^2}{\partial x_3^2} - k^2 \right) \hat{u}_2^n \right] \\ &\quad - ik_2 \left[ \left( \frac{3}{2} \hat{S}_1^n - \frac{1}{2} \hat{S}_1^{n-1} \right) + \frac{1}{2 \operatorname{Re}_\tau} \left( \frac{\partial^2}{\partial x_3^2} - k^2 \right) \hat{u}_1^n \right] + \\ &\quad + \frac{1}{2 \operatorname{Re}_\tau} \left( \frac{\partial^2}{\partial x_3^2} - k^2 \right) \omega_3^{n+1}\end{aligned}$$

Finally

$$\left( \frac{\partial^2}{\partial x_3^2} - k^2 \right) \omega_3^{n+1} = -\frac{1}{\gamma} (ik_1 H_2^n - ik_2 H_1^n)$$

$$H_i^n = \Delta t \left( \frac{3}{2} S_i^n - \frac{1}{2} S_i^{n-1} \right) + \left( \gamma \frac{\partial^2}{\partial x_3^2} + (1 - \gamma k^2) \right) \hat{u}_i^n$$

Similarly for the vertical velocity

$$\left( \frac{\partial^2}{\partial x_3^2} - \beta^2 \right) \left( \frac{\partial^2}{\partial x_3^2} - k^2 \right) \hat{u}_3^{n+1} = \frac{H^n}{\gamma}$$

$$H^n = \frac{\partial}{\partial x_3} (ik_1 H_1^n + ik_2 H_2^n) + k^2 H_3^n$$

$$\gamma = \frac{\Delta t}{2 \text{Re}_\tau}; \beta = \frac{1 + \gamma k^2}{\gamma}$$

Discrete “version” of the final system to be solved

$$\left( \frac{\partial^2}{\partial x_3^2} - k^2 \right) \omega_3^{n+1} = -\frac{1}{\gamma} (ik_1 H_2^n - ik_2 H_1^n)$$



HELMHOLTZ EQUATIONS

$$\left( \frac{\partial^2}{\partial x_3^2} - \beta^2 \right) \left( \frac{\partial^2}{\partial x_3^2} - k^2 \right) \hat{u}_3^{n+1} = \frac{H^n}{\gamma}$$

$$ik_1 \hat{u}_1 + ik_2 \hat{u}_2 + \frac{\partial}{\partial x_3} \hat{u}_3 = 0$$

$$\hat{\omega}_3 = ik_1 \hat{u}_2 - ik_2 \hat{u}_1$$

## CHEBYSHEV-TAU METHOD

$$\phi''(x) - \alpha^2 \phi(x) = H(x)$$

General Helmholtz equation

$$p_1\phi(-1) + q_1\phi'(-1) = r_1$$

Boundary conditions

$$p_2\phi(+1) + q_2\phi'(+1) = r_2$$

Functions are represented through Chebyshev polynomials

$$\phi(x) = \sum_{n=0}^N a_n T_n(x) = a_0 T_0(x) + a_1 T_1(x) + \dots$$

$$H(x) = \sum_{n=0}^N b_n T_n(x) = b_0 T_0(x) + b_1 T_1(x) + \dots$$

$$\phi''(x) - \alpha^2 \phi(x) = H(x) \quad \text{Integrated twice Helmholtz equation}$$

$$\phi(x) - \alpha^2 \iint \phi(y) dy dy = \iint H(y) dy dy + Ax + B$$

Using the Chebyshev representation of the functions, we have

$$\sum_{n=0}^N a_n T_n(x) - \alpha^2 \iint \sum_{n=0}^N a_n T_n(y) dy dy = \iint \sum_{n=0}^N b_n T_n(y) dy dy + Ax + B$$

Integrals can be evaluated using some properties of Chebyshev series

$$\int \phi(y) dy = \int \sum_{n=0}^N a_n T_n(y) dy = \sum_{n=1}^{N+1} l_n T_n(x)$$

$$\iint \phi(y) dy = \int \sum_{n=1}^{N+1} l_n T_n(y) dy = \sum_{n=2}^{N+2} m_n T_n(x)$$

Where the coefficients  $l_n$  and  $m_n$  can be obtained from  $a_n$  using recursive rules of Chebyshev polynomials

We finally get

$$\sum_{n=0}^N a_n T_n(x) - \alpha^2 \sum_{n=2}^{N+2} m_n T_n(x) = \sum_{n=2}^{N+2} f_n T_n(x) + A T_1(x) + B T_0(x)$$

where

$$T_1(x) = x$$

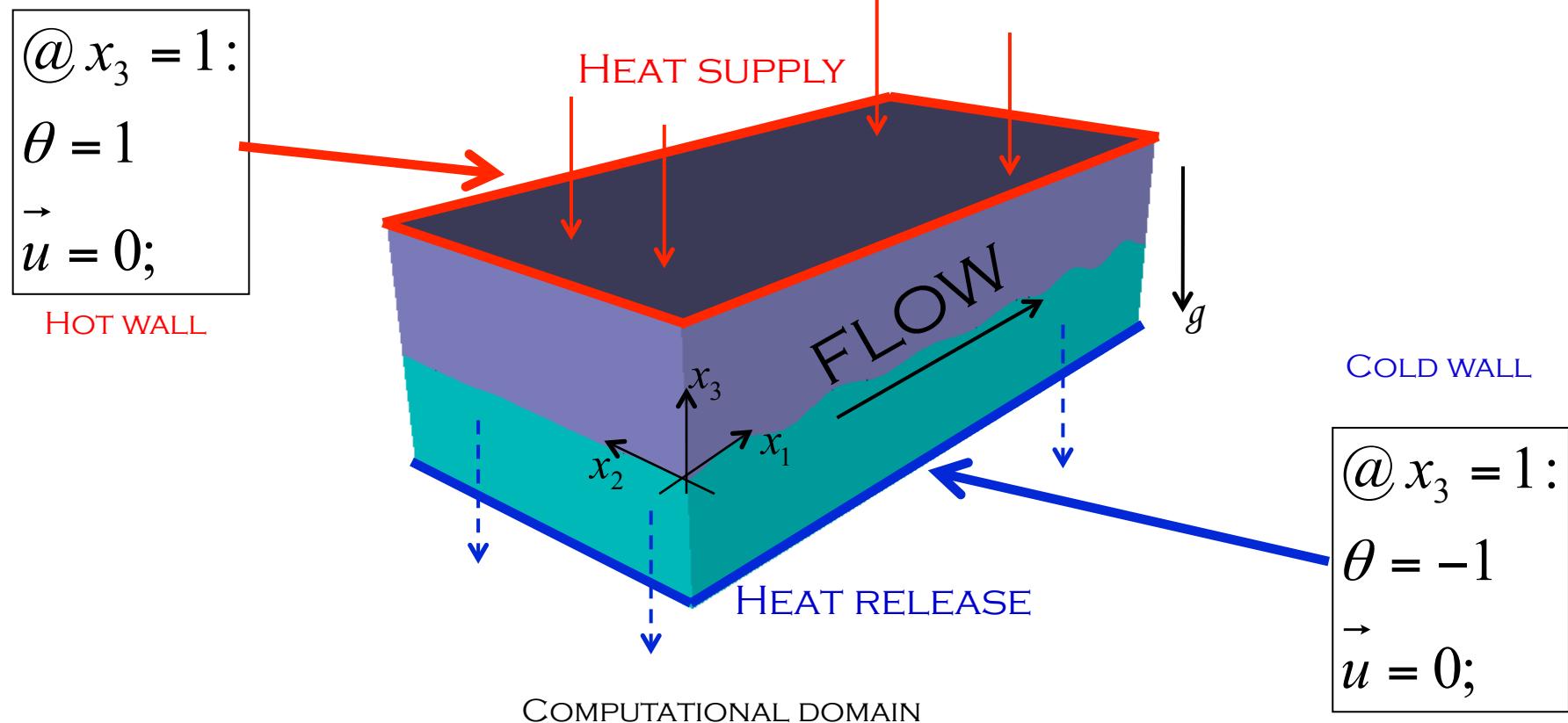
$$T_0(x) = 1$$

To solve this equation, we first define the residual:

$$R(x) = (a_0 - B)T_0(x) + (a_1 - A)T_1(x) + \sum_{n=2}^N s_n T_n(x)$$

Minimize  $R(x)$ ; solve a linear system (Gauss) to find the coefficients of the Chebyshev series  $a_n$

$$s_n = a_n - \alpha^2 m_n - f_n$$

PART II:  
STABLY STRATIFIED TURBULENCE

STABLY STRATIFIED TURBULENCE:  
OB APPROX.

$$\frac{\partial u_i}{\partial x_i} = 0$$

$$\frac{\partial u_i}{\partial t} + u_j \frac{\partial u_i}{\partial x_j} = \frac{1}{\text{Re}_\tau} \frac{\partial^2 u_i}{\partial x_j^2} - \frac{\partial p}{\partial x_i} - \frac{1}{16} \frac{Gr}{\text{Re}_\tau^2} \delta_{i,3} \theta$$

$$\frac{\partial \theta}{\partial t} + u_j \frac{\partial \theta}{\partial x_j} = \frac{1}{\text{Re}_\tau \text{Pr}} \frac{\partial^2 \theta}{\partial x_j^2}$$

## PHYSICAL PARAMETERS

$$\Delta\theta_{HC} = 40^\circ C$$

$$\theta_m = 50^\circ C$$

$$\text{Pr} = 3$$

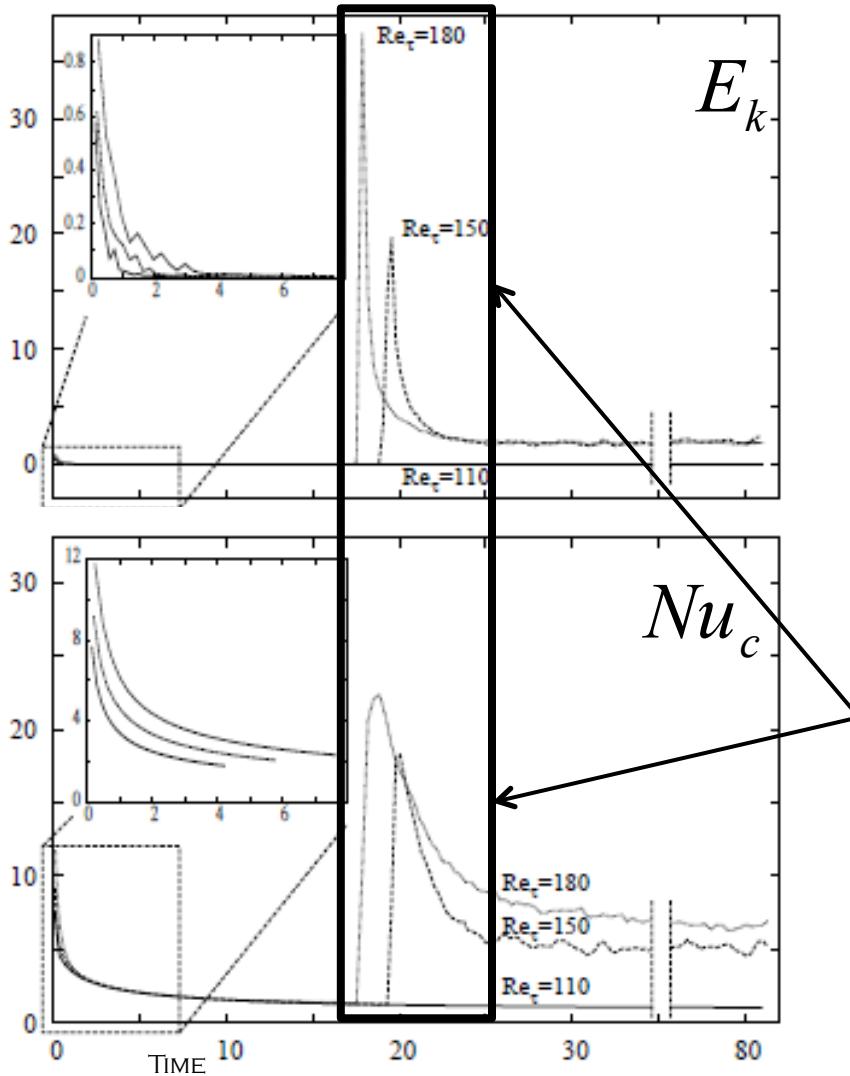
$$\text{Re}_\tau = 110 \div 180$$

$$Gr \approx 10^7$$

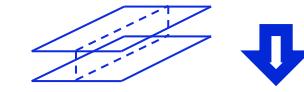
$$Ri_\tau = \frac{Gr}{\text{Re}_\tau^2}$$

$$Gr = \frac{g \beta \Delta \theta_{HC} (2h)^3}{\nu^2} \quad \text{Pr} = \frac{\mu c_p}{\lambda} = \frac{\nu}{k}$$

# STABLY STRATIFIED TURBULENCE: TRANSIENT DEVELOPMENT OF THE FLOW

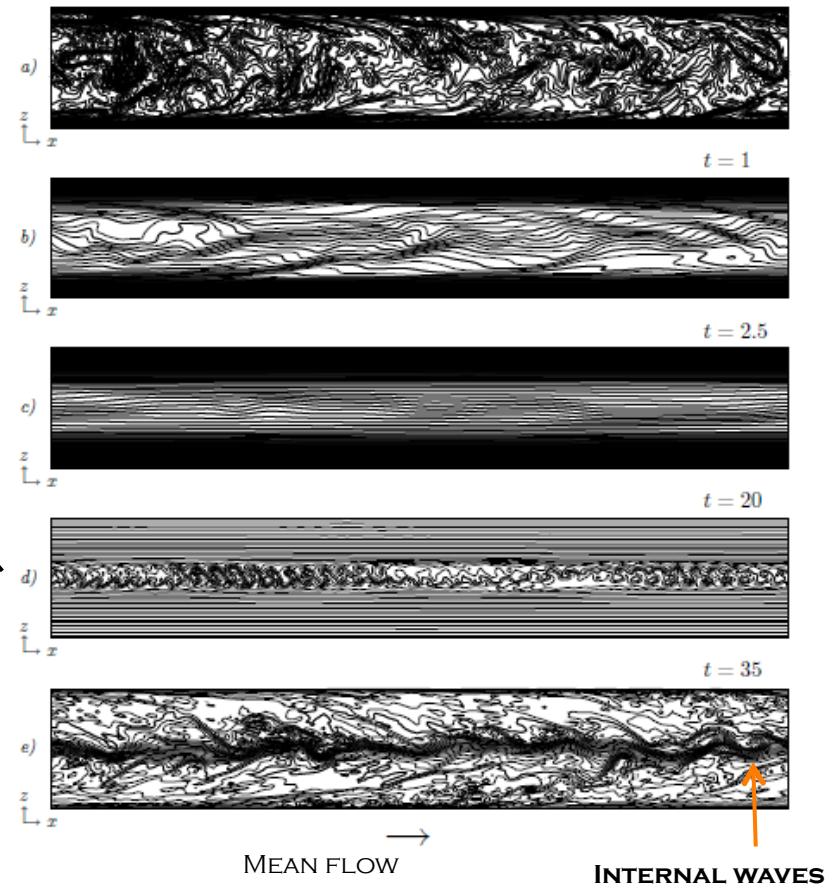


TEMPERATURE FIELD  
(LONGITUDINAL SECTION)



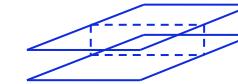
$t = 0$

TIME

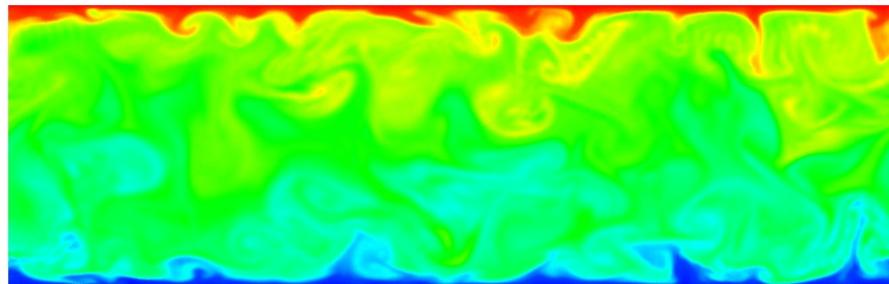


# THE ROLE OF BUOYANCY: NEUTRALLY-BUOYANT VS STABLY STRATIFIED

VISUALIZATION ON A CROSS SECTION

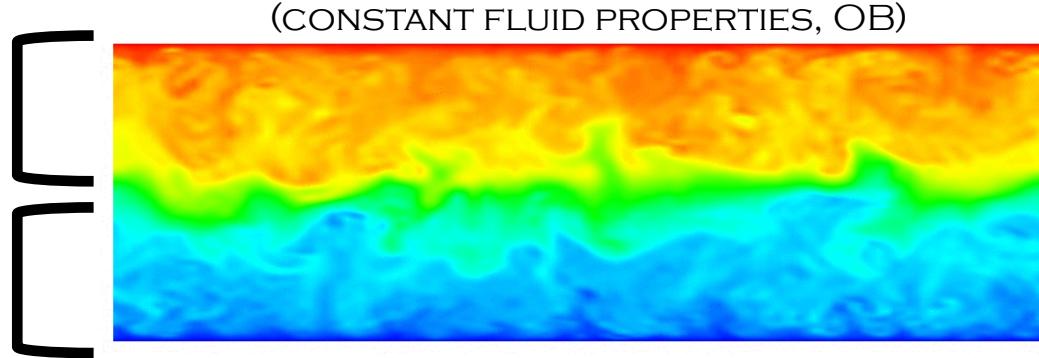


NEUTRALLY BUOYANT



MIXING  
(DRIVEN BY TURBULENCE  
STRUCTURES)

INTERNAL WAVES=BARRIER  
(TWO "INDEPENDENT" ZONES)

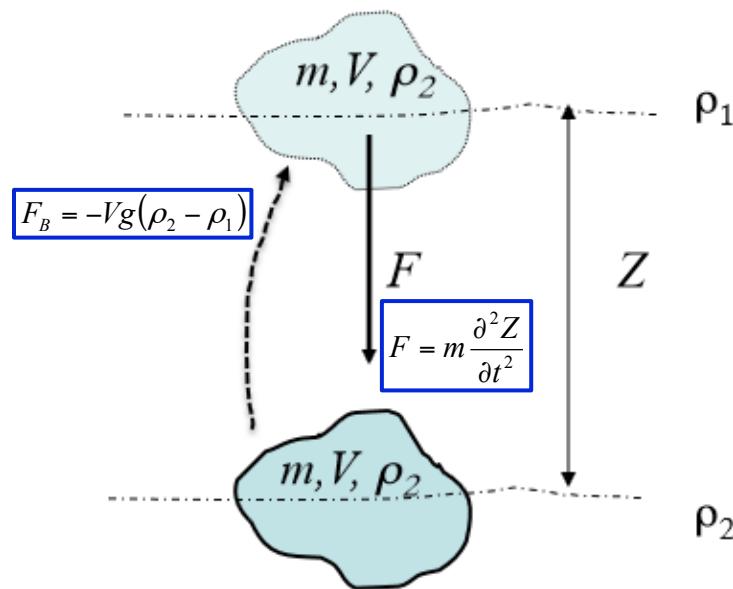
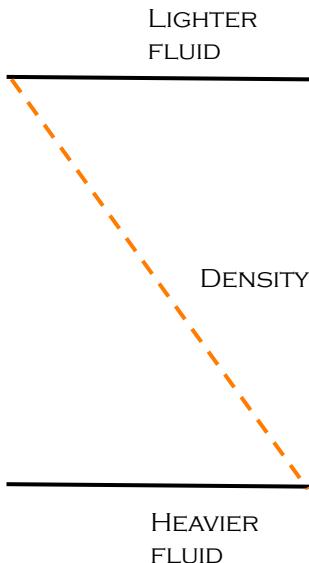


STABLY-STRATIFIED  
(CONSTANT FLUID PROPERTIES, OB)

THE ORING OF INTERNAL WAVES...NEXT SLIDE!

STABLY STRATIFIED TURBULENCE:  
THE ORIGIN OF INTERNAL WAVES

FORCE BALANCE ON A FLUID PARTICLE:



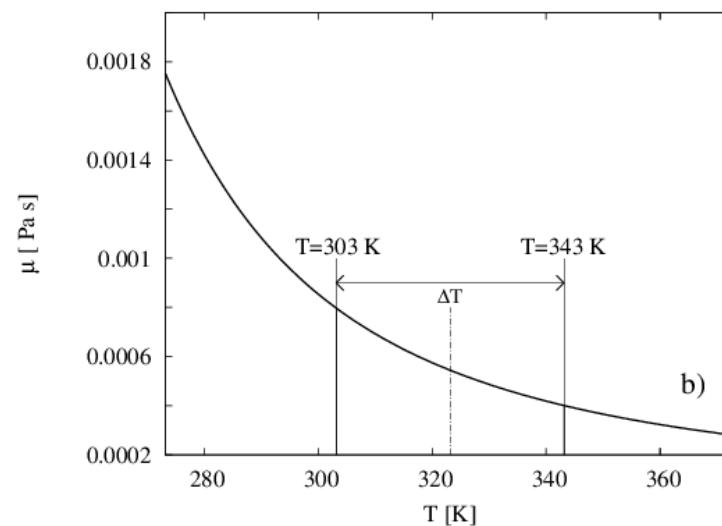
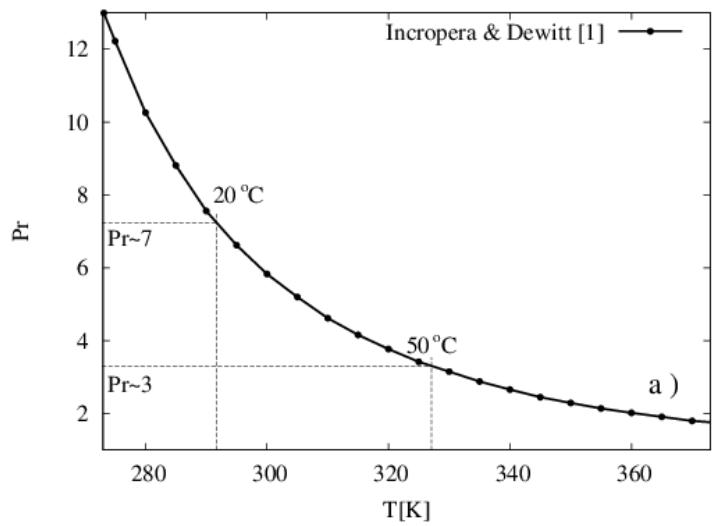
$$\begin{cases} \frac{\partial^2 Z}{\partial t^2} = \frac{g}{\rho} \frac{\partial \rho}{\partial z} Z = N^2 Z \\ N = \left( \frac{g}{\rho} \frac{\partial \rho}{\partial z} \right)^{1/2} \end{cases}$$

N: **BUOYANCY FREQUENCY**  
OR  
**BRUNT-VAISSALA FREQUENCY**

# THERMAL-STRATIFICATION IN WATER BEYOND THE BOUSSINESQ ASSUMPTION

So far we have considered uniform fluid properties

*However...*



$\mu, \text{Pr}$

# STABLY STRATIFIED TURBULENCE UNDER NOB CONDITIONS: EQUATIONS

$$\frac{\partial u_i}{\partial x_i} = 0$$

$$\frac{\partial u_i}{\partial t} = S_i + \frac{1}{Re_\tau} \frac{\partial^2 u_i}{\partial x_j^2} - \frac{\partial p}{\partial x_i}$$

$$\frac{\partial \theta}{\partial t} + u_j \frac{\partial \theta}{\partial x_j} = \frac{1}{Re_\tau \cdot Pr} \frac{\partial^2 \theta}{\partial x_j^2}$$

$$\Delta\theta = 40^\circ C$$

$$\theta_m = 50^\circ C$$

$$Pr = 3$$

$$Gr \approx 10^7$$

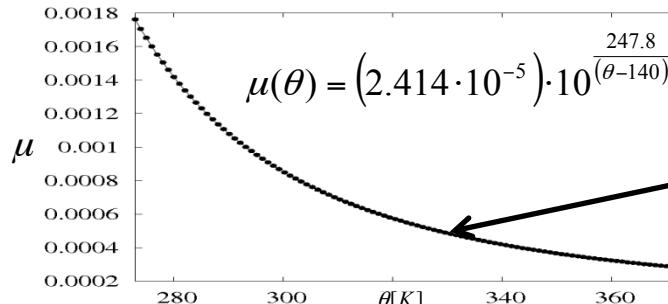
$$Ri_\tau = \frac{Gr}{Re_\tau^2}$$

$$S_i = -u_j \frac{\partial u_i}{\partial x_j} + \delta_{i,1} + \delta_{i,3} \frac{1}{16} \frac{Gr}{Re_\tau^2} \theta$$

UNIFORM THERMOPHYSICAL PROPERTIES  
(OB)

$$S_i = -u_j \frac{\partial u_i}{\partial x_j} + \frac{1}{Re_\tau} \frac{\partial}{\partial x_j} \left( \mu_v \frac{\partial u_i}{\partial x_j} \right) + \delta_{i,1} + \delta_{i,3} \frac{1}{16} \frac{Gr}{Re_\tau^2} \theta$$

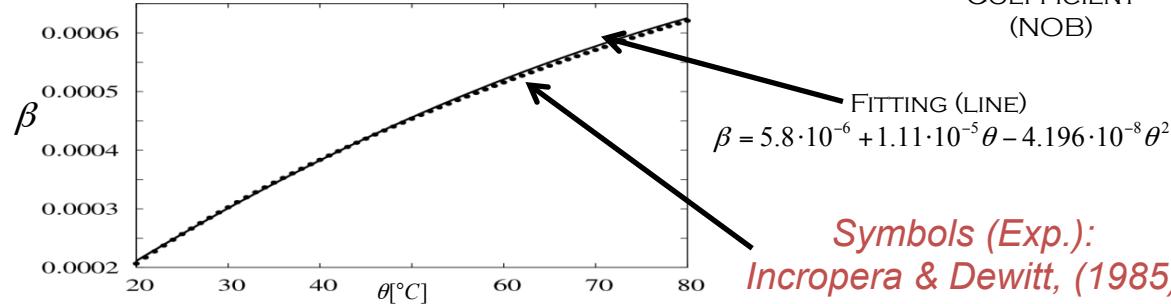
TEMPERATURE-  
DEPENDENT  
VISCOSITY (NOB)



Symbols (Exp.):  
*Incropera & Dewitt, (1985)*

$$S_i = -u_j \frac{\partial u_i}{\partial x_j} + \delta_{i,1} + \delta_{i,3} \cdot g_i \cdot \rho_{ref} \left( \exp \left( - \int_{\theta_{ref}}^{\theta} \beta(\theta) d\theta \right) - 1 \right)$$

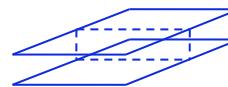
TEMPERATURE-  
DEPENDENT  
THERMAL  
EXPANSION  
COEFFICIENT  
(NOB)



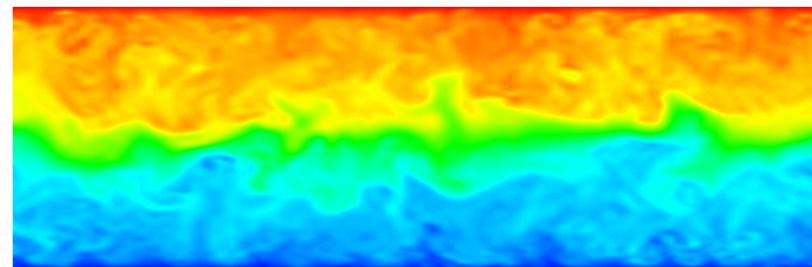
FITTING (LINE)  
Symbols (Exp.):  
*Incropera & Dewitt, (1985)*

# THE ROLE OF TEMPERATURE DEPENDENT FLUID PROPERTIES

VISUALIZATION ON A CROSS SECTION

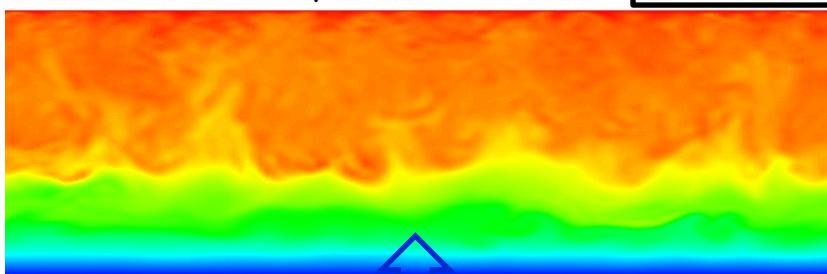


STABLY-STRATIFIED  
 $\beta, \mu = \text{CONSTANT - OB}$



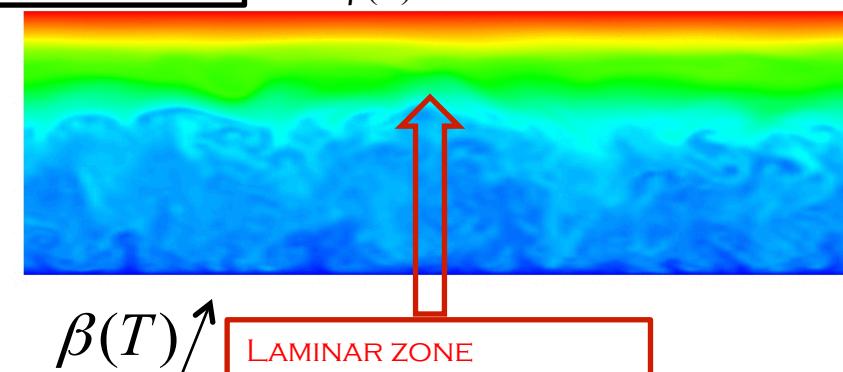
LOCAL DYNAMICS OF THE  
FLOW: VERY IMPORTANT  
(ONE-SIDED TURBULENCE)

STABLY-STRATIFIED  
 $\mu(T) - \text{NOB}$



$\mu(T) \uparrow$   
LAMINAR ZONE

STABLY-STRATIFIED  
 $\beta(T) - \text{NOB}$



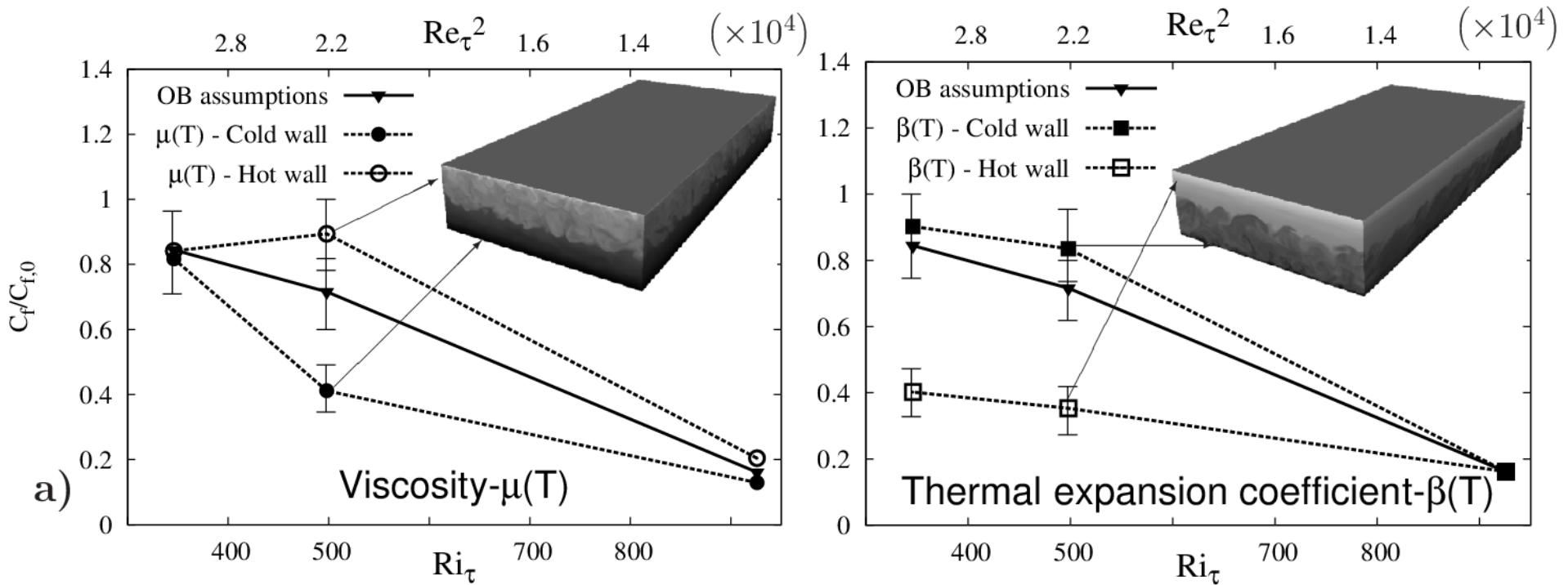
$\beta(T) \uparrow$   
LAMINAR ZONE

Zonta et al., JFM, 697, 175-203 (2012)

MOMENTUM AND  
HEAT TRANSFER COEFFICIENT-2

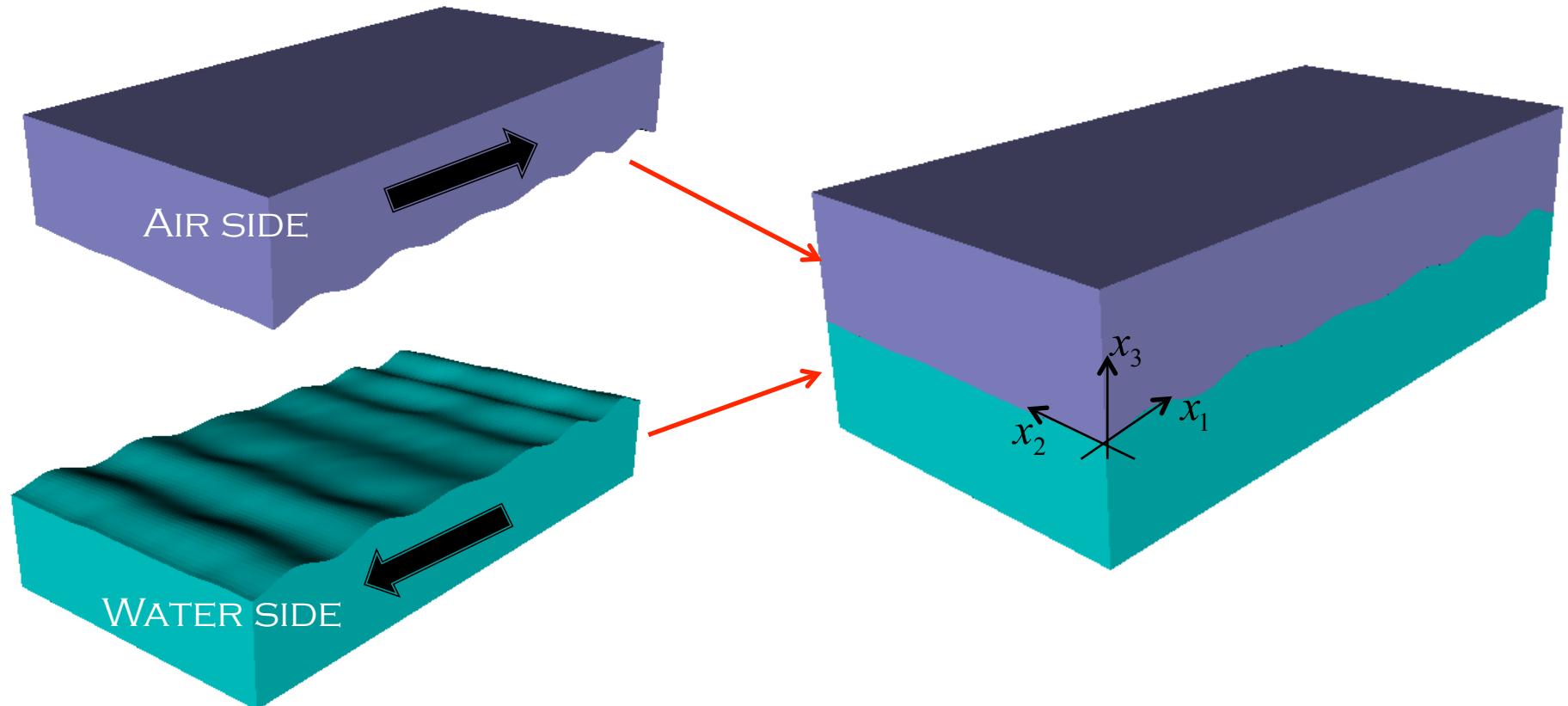
$$C_f = \frac{2\tau_w}{\rho u_b^2}$$

←  
LOCAL VALUE OF THE  
SHEAR STRESS



PART III:  
SURFACE WAVES

## PHYSICAL CONFIGURATION



COUNTERCURRENT AIR-WATER FLOW  
(DRIVEN BY PRESSURE GRADIENT)

GOVERNING EQUATIONS - PHYSICAL DOMAIN

$$\nabla \cdot \vec{u} = 0$$

$$\frac{\partial \vec{u}}{\partial t} + \vec{u} \cdot \nabla \vec{u} = -\nabla p + \frac{1}{\text{Re}} \nabla^2 \vec{u}$$

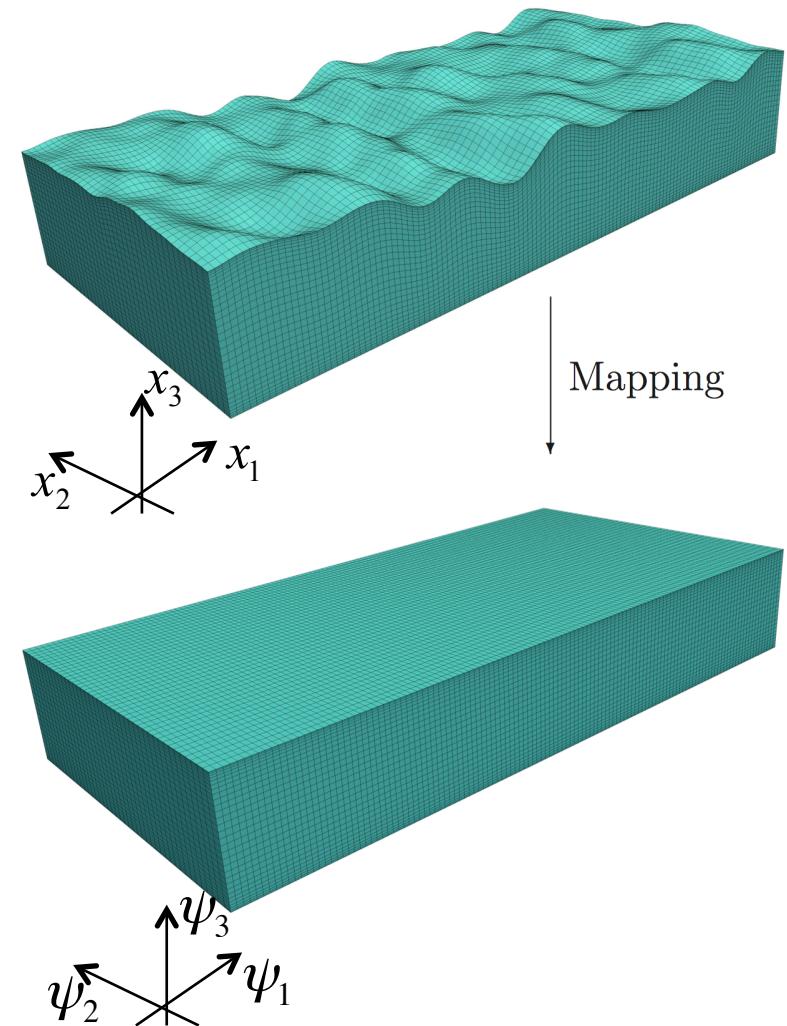
ALGEBRAIC MAPPING

$$\psi_1 = x$$

$$\psi_2 = y$$

$$\psi_3 = \frac{z}{h + \eta(x, y, t)}$$

$$\tau = t$$



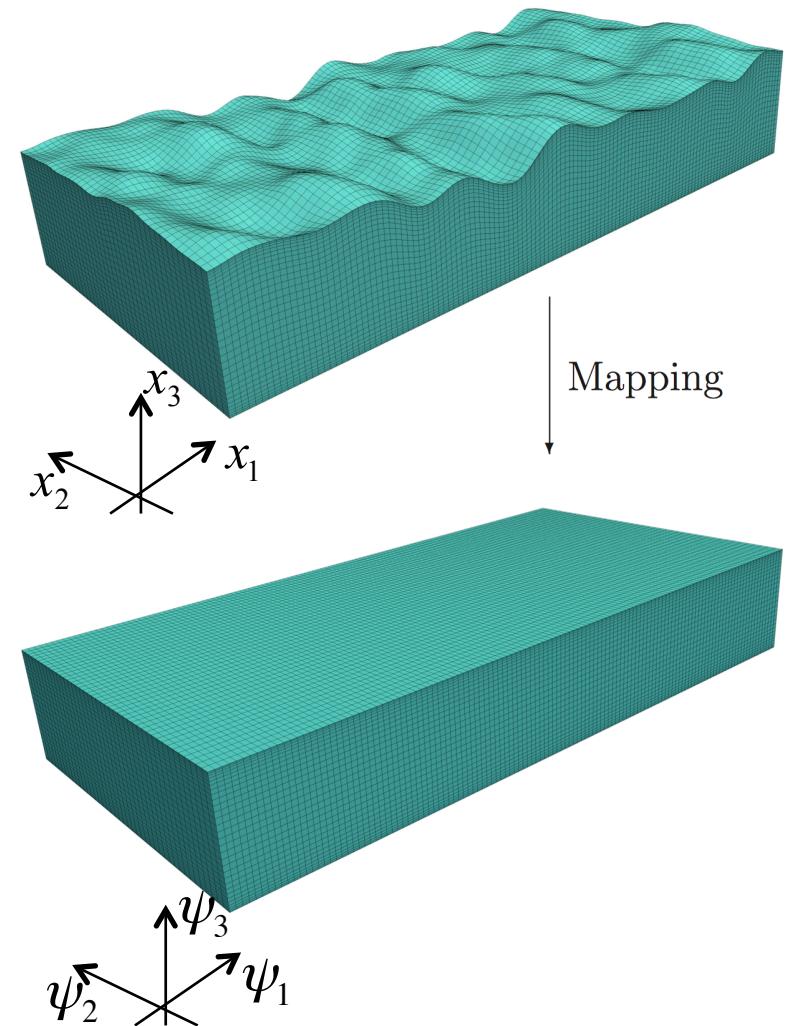
JACOBIAN & DERIVATIVES

$$J = \frac{\partial \psi}{\partial X}$$

$$\partial_X = J \cdot \partial_\psi$$

$$\partial_X = (\partial / \partial x, \partial / \partial y, \partial / \partial z)^T$$

$$\partial_\psi = (\partial / \partial \psi_1, \partial / \partial \psi_2, \partial / \partial \psi_3)^T$$



## FRACTIONAL-STEP TECHNIQUE

PROVISIONAL TIME STEP

$$\frac{\tilde{u} - u^n}{\Delta t} + \sum_{q=0}^{M-1} \alpha_q \nabla \cdot (u u)^{n-q} - \frac{1}{2 \text{Re}_\tau} \nabla^2 (\tilde{u} - u^n) = 0$$

CONVECTIVE TERM: ADAMS-BASHPFORT EXPLICIT ( $M=2$ ,  $\alpha_0=3/2$ ,  $\alpha_1=-1/2$ )

CORRECTION TO OBTAIN DIVERGENCE-FREE FIELD

$$\frac{u^{n+1} - \tilde{u}}{\Delta t} + \nabla p^{n+1} = 0$$

TAKING DIVERGENCE &  $U^{N+1}$  DIV. FREE:

$$\nabla^2 p^{n+1} = \frac{1}{\Delta t} \nabla \cdot \tilde{u}$$

SOLVE TO GIVE  $P^{N+1}$  &  $U^{N+1}$

## SPATIAL DISCRETIZATION

$$\Phi(\psi_i) = \sum_{k_1, k_2, k_3} \hat{\Phi}(k_1, k_2, n_3) e^{i(k_1\psi_1 + k_2\psi_2)} T_{n_3}(\psi_3)$$

SPECTRAL AMPLITUDE

CHEBYSHEV POLYNOMIALS

$$T_{n_3}(\psi_3) = \cos[n_3 \cos^{-1}(\psi_3 / h)]$$

DISCRETIZED  
GOVERNING EQ.:  
CHEB-TAU METHOD  
WITH PROPER B.C.

$$\left( \frac{d^2}{d\psi_3^2} - \beta \right) \hat{\tilde{u}}_i = \frac{\hat{H}_i}{\delta} - F\left(\nabla_{off}^2 u_i^n\right)$$

$$\hat{H}_i = \Delta t \left( \frac{3}{2} \hat{S}_i^n - \frac{1}{2} \hat{S}_i^{n-1} \right) + F\left(\delta \nabla_{diag}^2 u_i^n\right) + u_i^n$$

$$\beta = \frac{1 + \delta k^2}{\delta}; \quad \delta = \frac{\Delta t}{2 \operatorname{Re}_\tau}$$

$S_i^n$  (Convective terms)  
F (Four.-Cheb. Transform)

## BOUNDARY CONDITIONS

INTERFACE

KINEMATIC B.C.

$$\frac{\partial \eta}{\partial t} + u_1 \frac{\partial \eta}{\partial x_1} + u_2 \frac{\partial \eta}{\partial x_2} = u_3$$

DYNAMIC B.C.

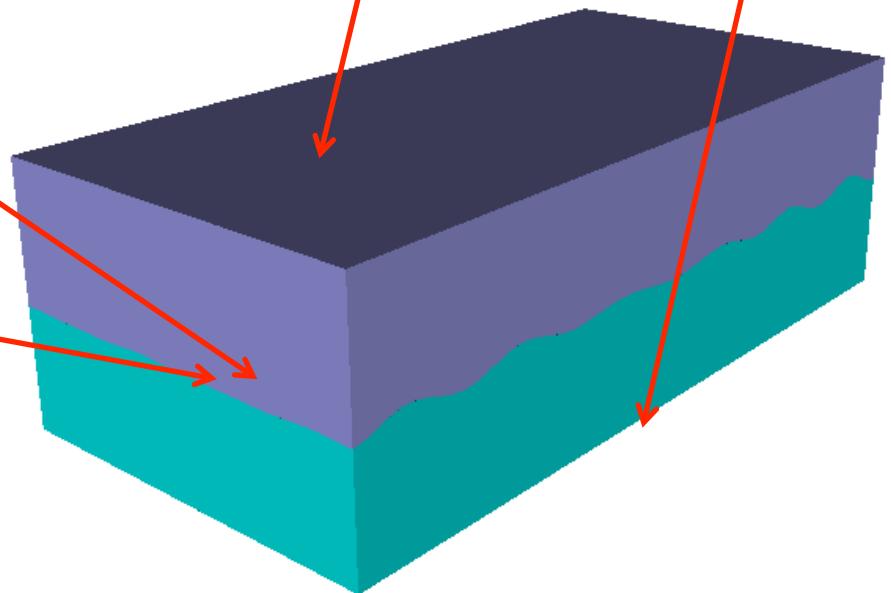
$$@x_3 = 0 : p_g - p_l - g(\rho_l - \rho_g)\eta = (\vec{\tau}_g - \vec{\tau}_l) \cdot \vec{n} + \gamma\kappa; \\ ((\vec{\tau}_l - \vec{\tau}_g) \cdot \vec{n}) \cdot \vec{t}_{1,2} = 0; \\ \vec{u}_g = \vec{u}_l;$$

*(CONTINUITY OF STRESS AND VELOCITY)*

FREE-SLIP BC

$$@x_3 = \pm h :$$

$$\frac{\partial u_{1,2}}{\partial x_3} = \frac{\partial p}{\partial x_3} = 0; u_3 = 0;$$



# SIMULATIONS: PLAN OF EXPERIMENTS

*PHYSICAL PROBLEM*

*LIQUID (l) : WATER*

*GAS (g): AIR*

*PHYSICAL PARAMETERS*

$$Re_{\tau,g} = \frac{\rho_g u_{\tau,g} h_g}{\mu_g}$$

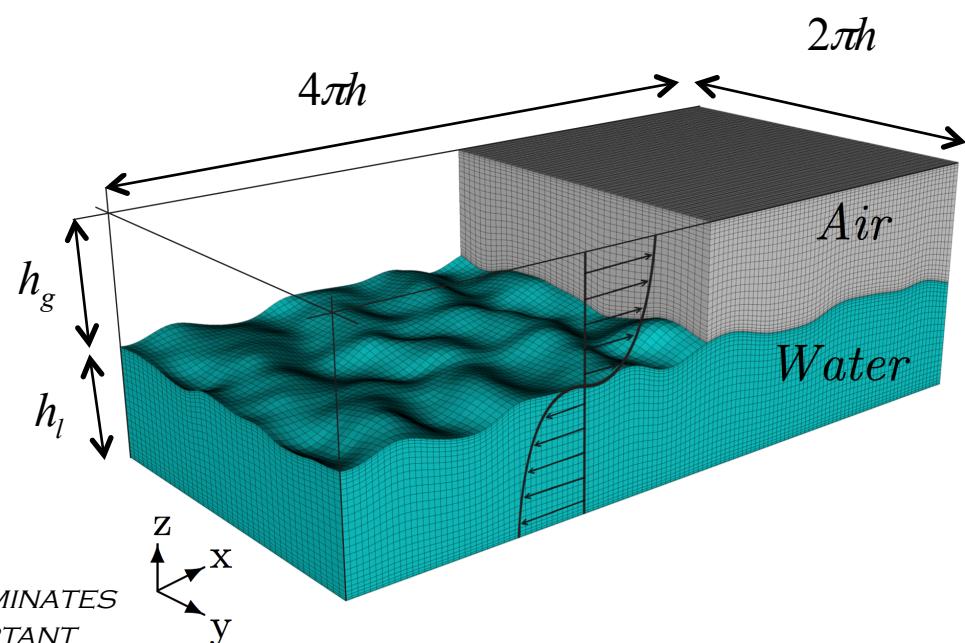
$$Re_{\tau,l} = \frac{\rho_l u_{\tau,l} h_l}{\mu_l}$$

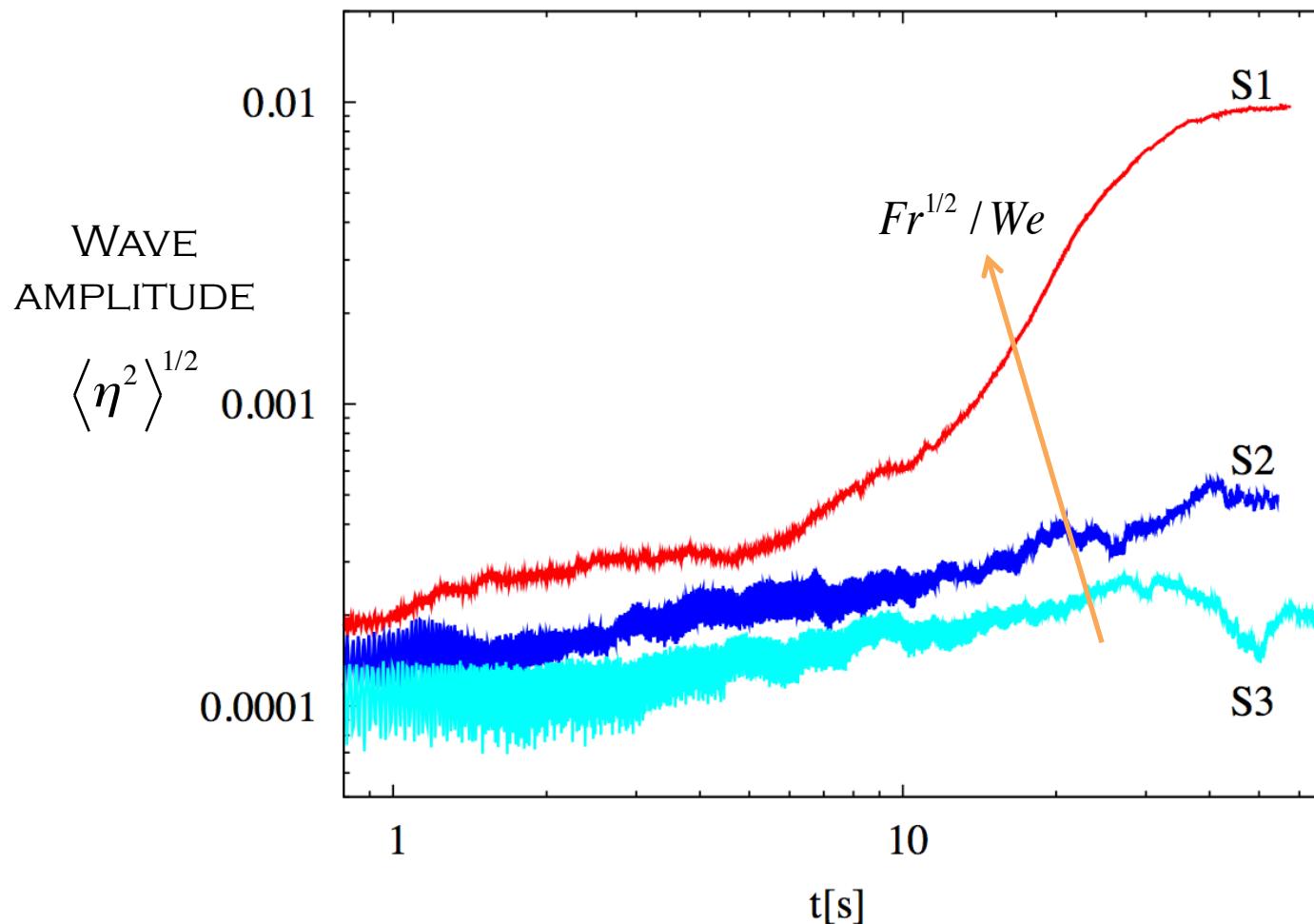
$$We = \frac{\rho_l h_l u_{\tau,l}^2}{\gamma} \quad \xrightarrow{\text{SURFACE TENSION}}$$

$$Fr = \frac{\rho_l u_{\tau,l}^2}{gh_l(\rho_l - \rho_g)}$$

$Fr^{1/2} / We$  → SMALLER → SURF. TENSION DOMINATES  
LARGER → GRAVITY MORE IMPORTANT

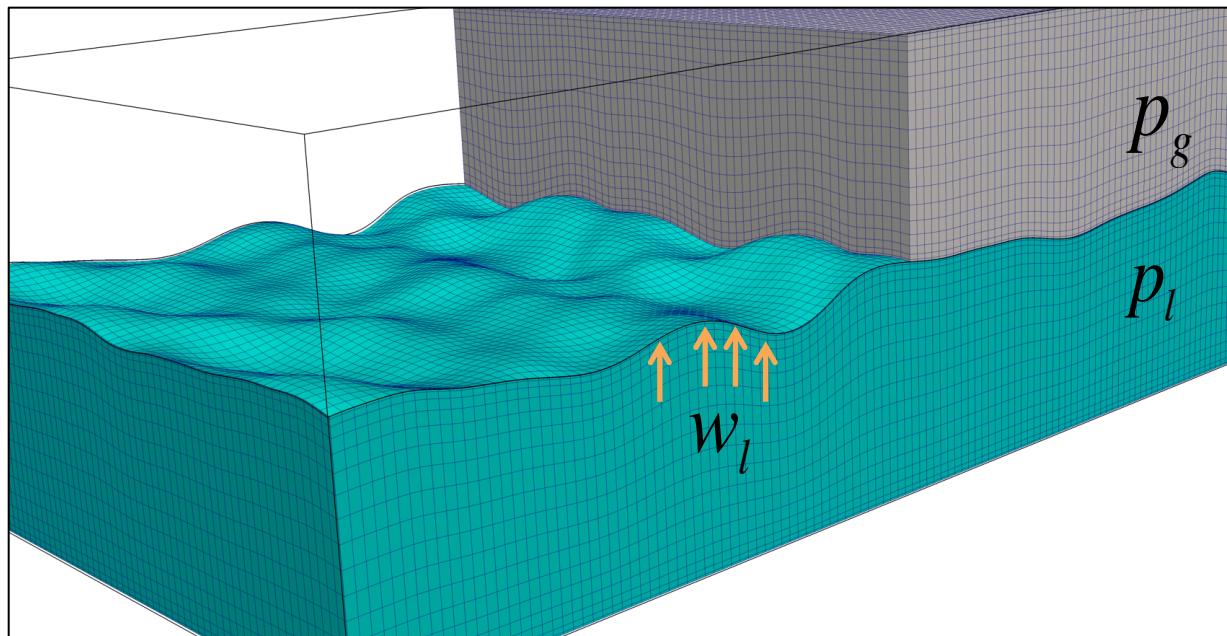
Simulation	h	Re <sub>τ</sub>	We	Fr	Fr <sup>1/2</sup> /We
S1	0.045	170	8.5x10 <sup>-4</sup>	2.9x10 <sup>-6</sup>	2.03
S2	0.05	170	7.6x10 <sup>-4</sup>	2.2x10 <sup>-6</sup>	1.93
S3	0.06	170	6.3x10 <sup>-4</sup>	1.3x10 <sup>-6</sup>	1.4



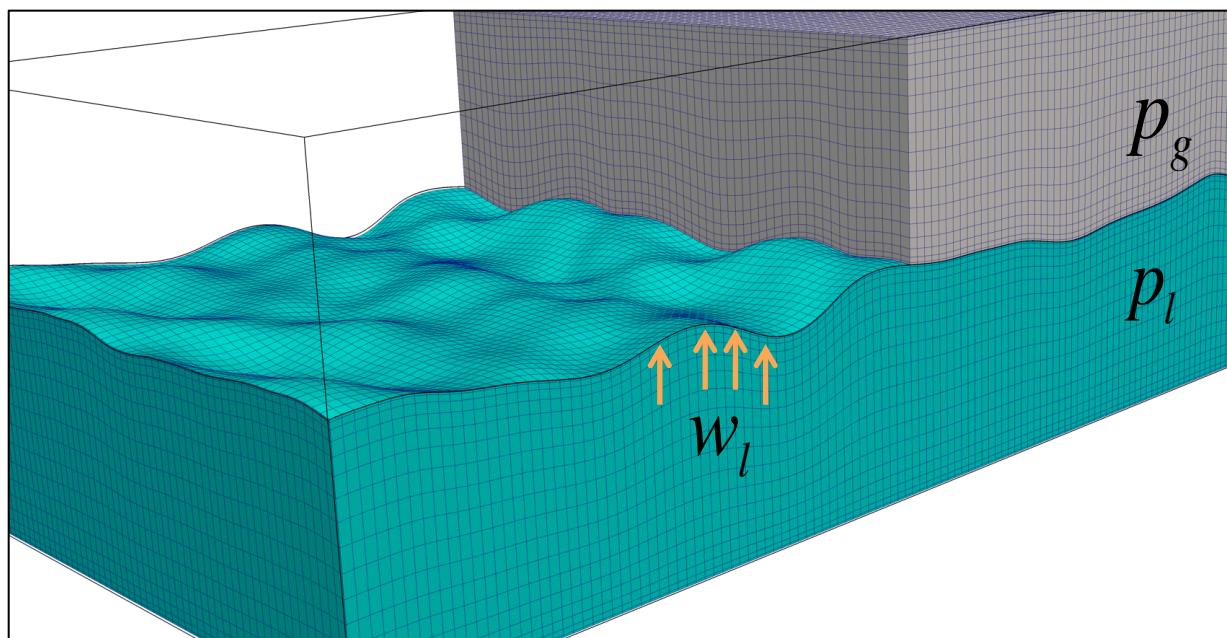


HOW TO EXPLAIN THE TIME BEHAVIOR OF WAVES?

## CONSIDER THE ORIGIN OF A WAVE...



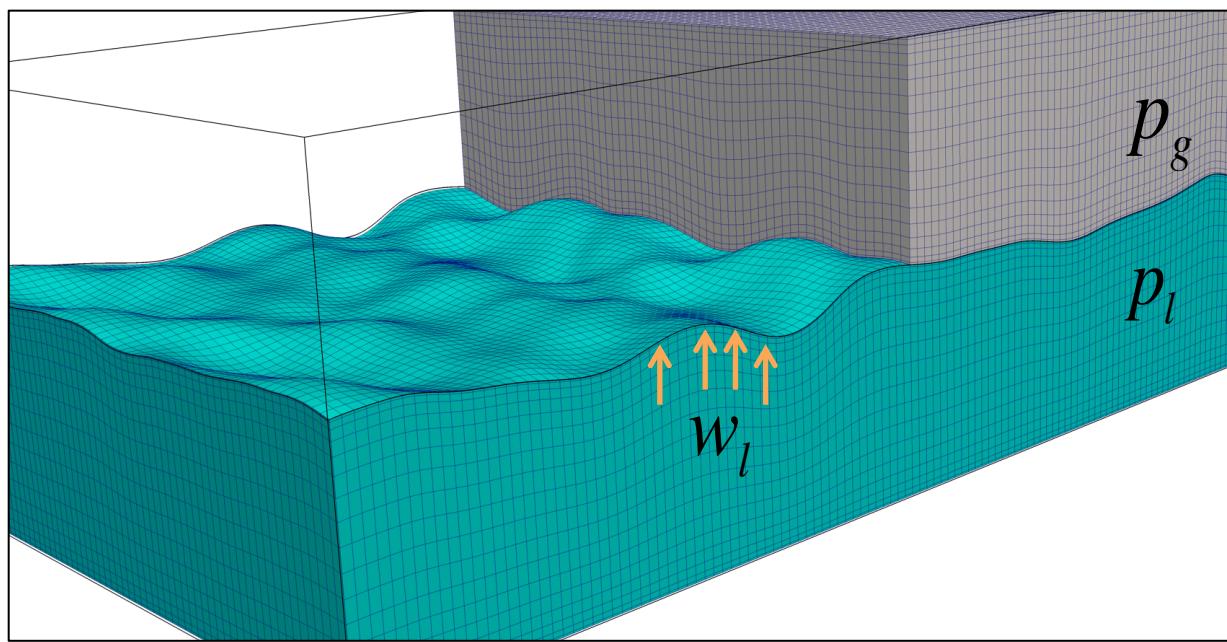
## CONSIDER THE ORIGIN OF A WAVE...



WE EXPRESS THE VARIATION  
OF THE INTERFACE AREA:  
(Hoepffner et al., PRL 2011)

$$\frac{dA}{dt} \propto h w_l$$

## CONSIDER THE ORIGIN OF A WAVE...

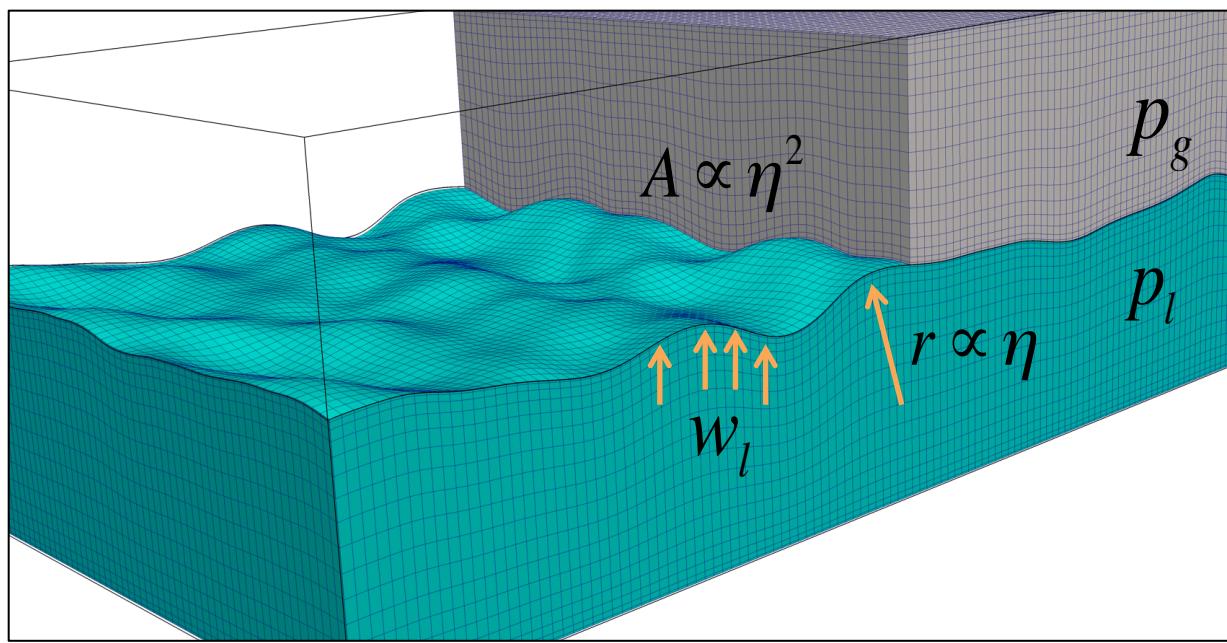


WE EXPRESS THE VARIATION  
OF THE INTERFACE AREA:  
(Hoepffner et al., PRL 2011):

$$\frac{dA}{dt} \propto h w_l$$

$$\left\{ \begin{array}{l} \Delta p \propto \rho w_l^2 \\ \Delta p = \frac{\gamma}{r} \end{array} \right. \xrightarrow{\text{SURFACE TENSION}} w_l \propto r^{-1/2}$$

## CONSIDER THE ORIGIN OF A WAVE...

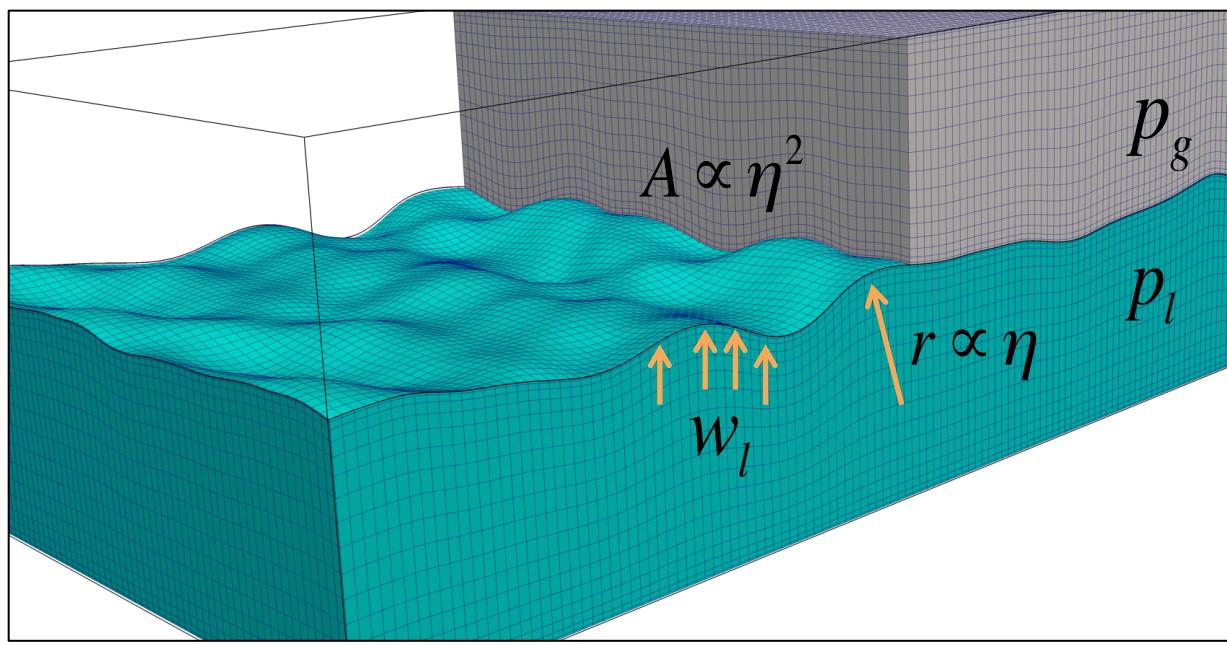


WE EXPRESS THE VARIATION  
OF THE INTERFACE AREA:  
(Hoepffner et al., PRL 2011):

$$\frac{dA}{dt} \propto h w_l$$

$$\left\{ \begin{array}{l} \Delta p \propto \rho w_l^2 \\ \Delta p = \frac{\gamma}{r} \end{array} \right. \xrightarrow{\text{SURFACE TENSION}} w_l \propto r^{-1/2}$$

## CONSIDER THE ORIGIN OF A WAVE...



WE EXPRESS THE VARIATION OF THE INTERFACE AREA:  
(Hoepffner et al., PRL 2011):

$$\frac{dA}{dt} \propto h w_l$$

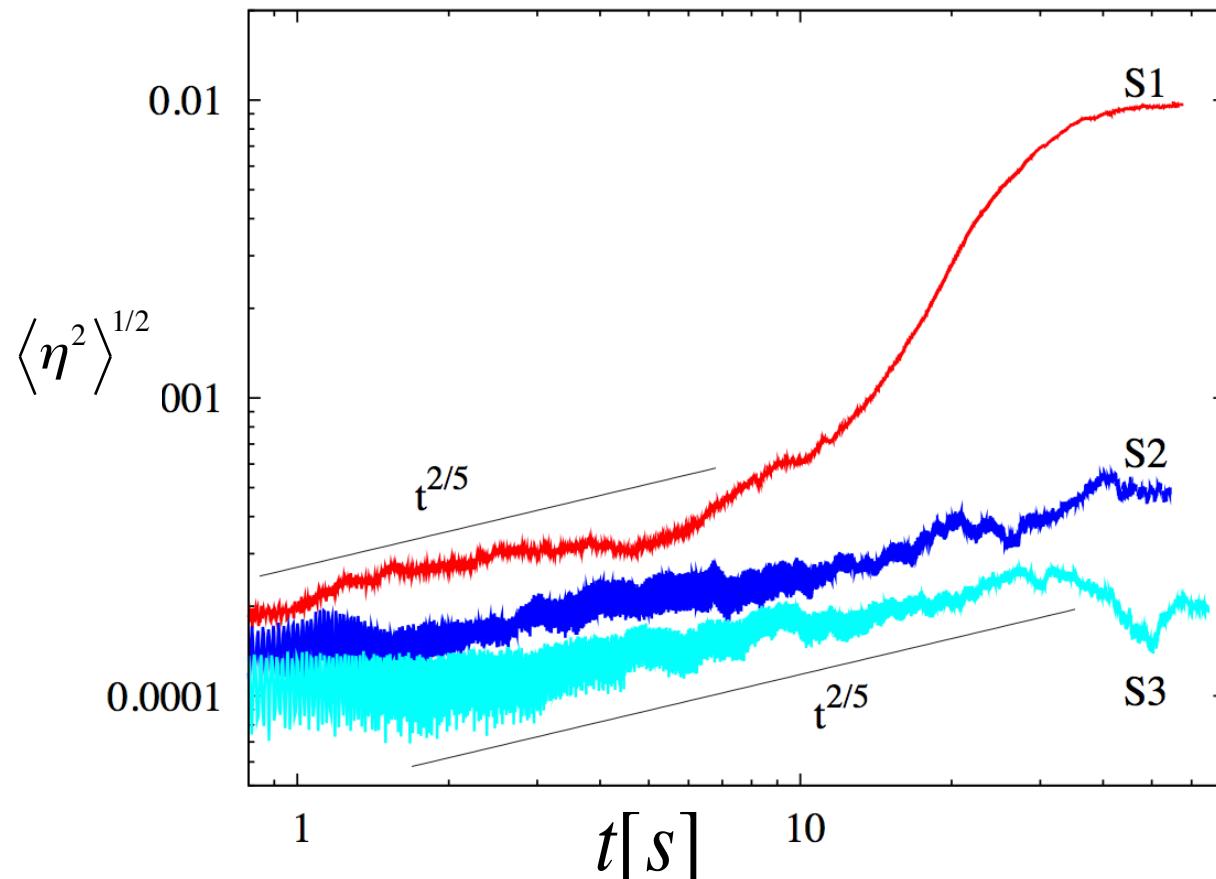
$$\left\{ \begin{array}{l} \Delta p \propto \rho w_l^2 \\ \Delta p = \frac{\gamma}{r} \end{array} \right. \xrightarrow{\text{SURFACE TENSION}} w_l \propto r^{-1/2}$$

AFTER SOME ALGEBRA:

$$\frac{d\eta^2}{dt} \propto \frac{1}{\eta^{1/2}} \longrightarrow \eta \propto t^{2/5}$$

TRANSIENT GROWTH OF WAVES:  
DNS VS SIMPLIFIED MODEL

## WAVE AMPLITUDE OVER TIME



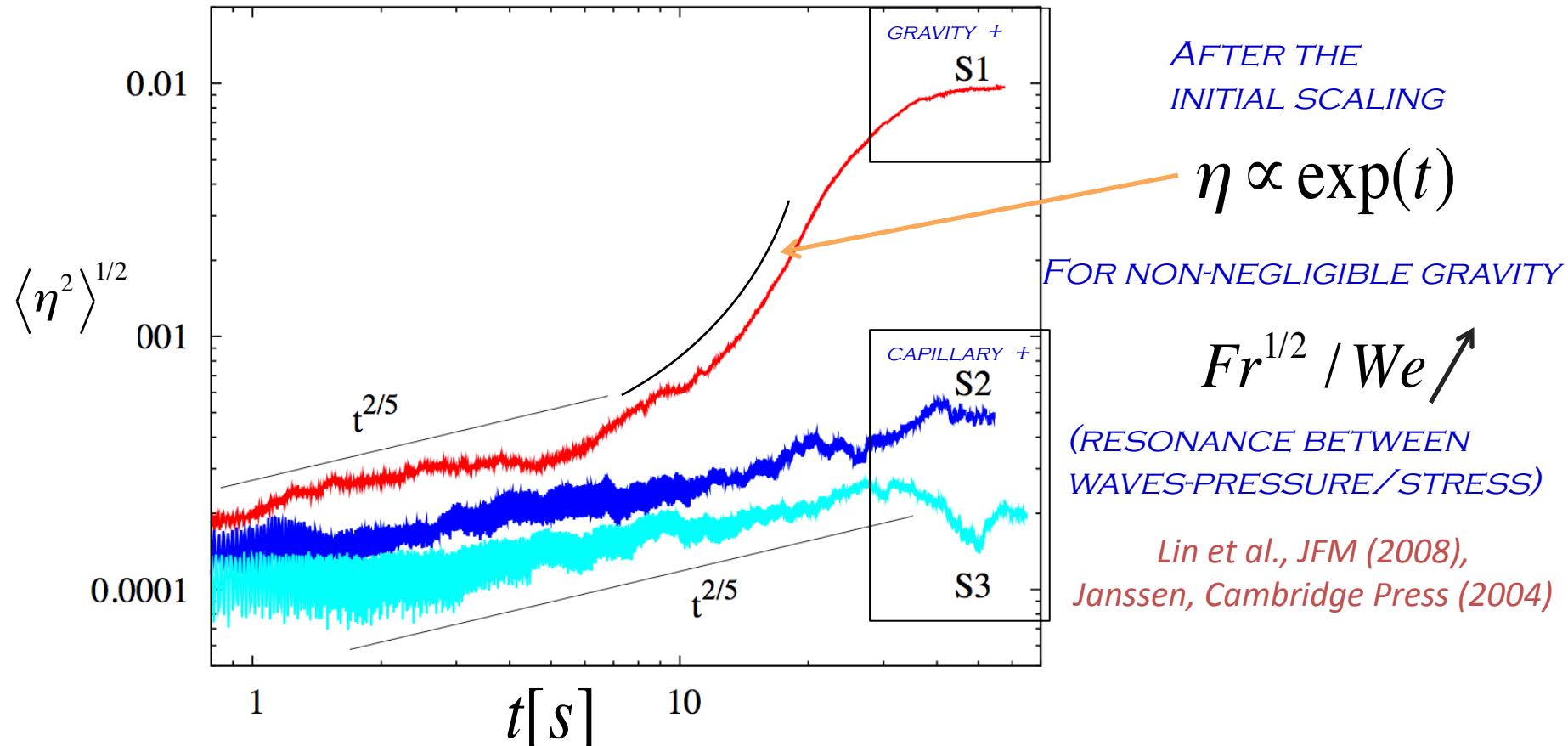
*INITIAL SCALING*

$$\eta \propto t^{2/5}$$

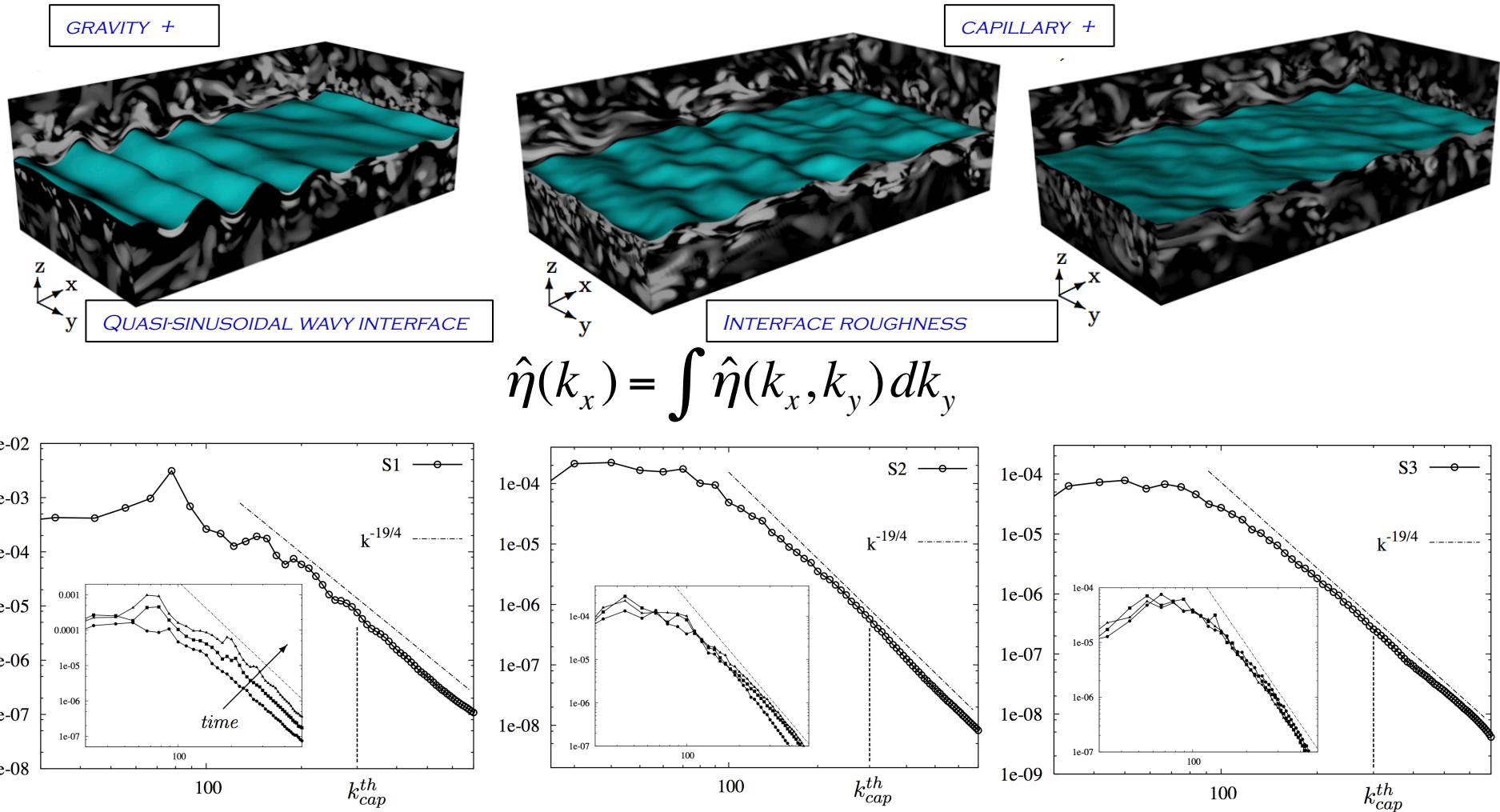
*QUITE "ROBUST" WITHIN THE RANGE OF PARAMETERS INVESTIGATED*

*WHATEVER THE PHYSICAL PARAMETERS, CAPILLARITY DOMINATES AT THE BEGINNING*

*Zonta et al., JFM (2015)*

TRANSIENT GROWTH OF WAVES:  
SMALL WAVES GROW

# THE STRUCTURE OF THE INTERFACE DEFORMATION

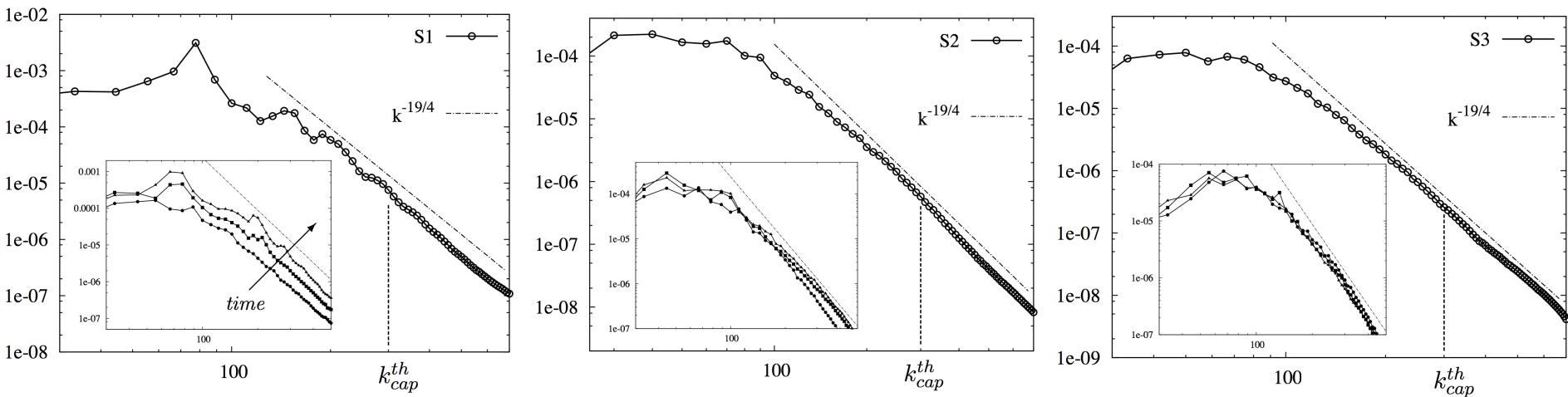


CFR. WITH WAVE TURBULENCE THEORY (Pushkarev & Zakharov, PRL 1996; Falcon et al., PRL 2007)

HPC methods for Computational Fluid Dynamics and Astrophysics

CINECA, Bologna, 2-4 Nov. 2016

# THE STRUCTURE OF THE INTERFACE DEFORMATION

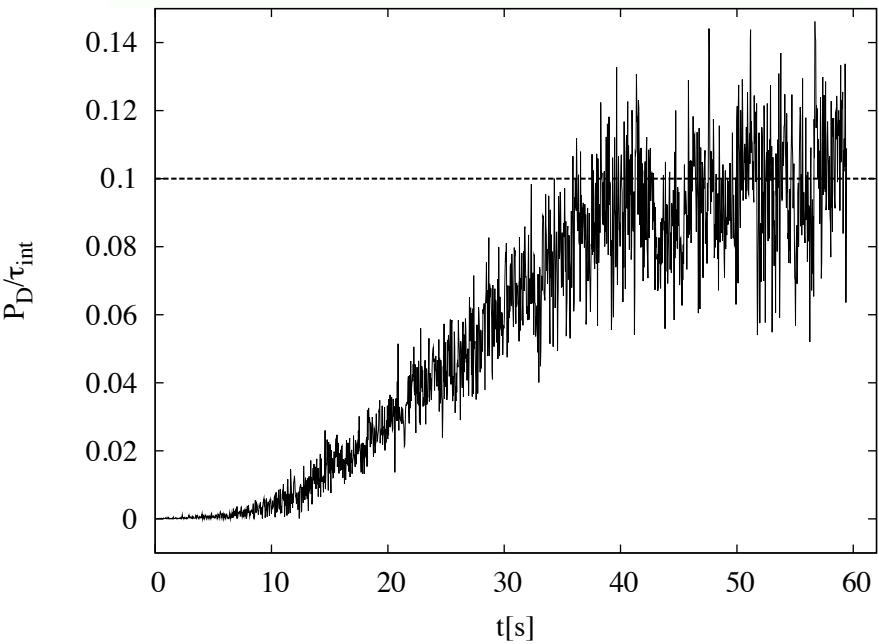
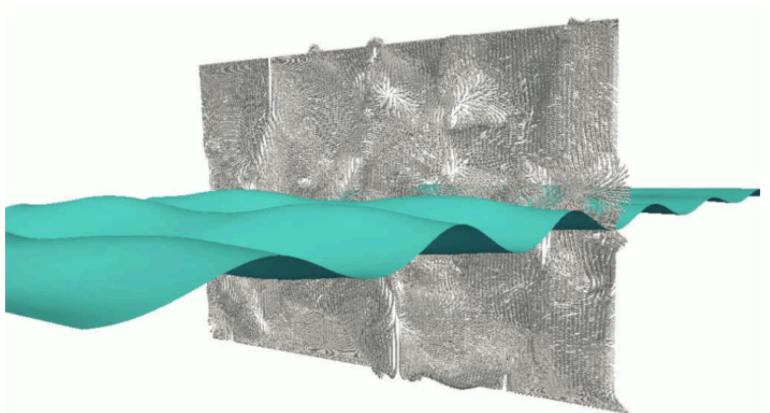


CFR. WITH WAVE TURBULENCE THEORY (Pushkarev & Zakharov, PRL 1996; Falcon et al., PRL 2007)

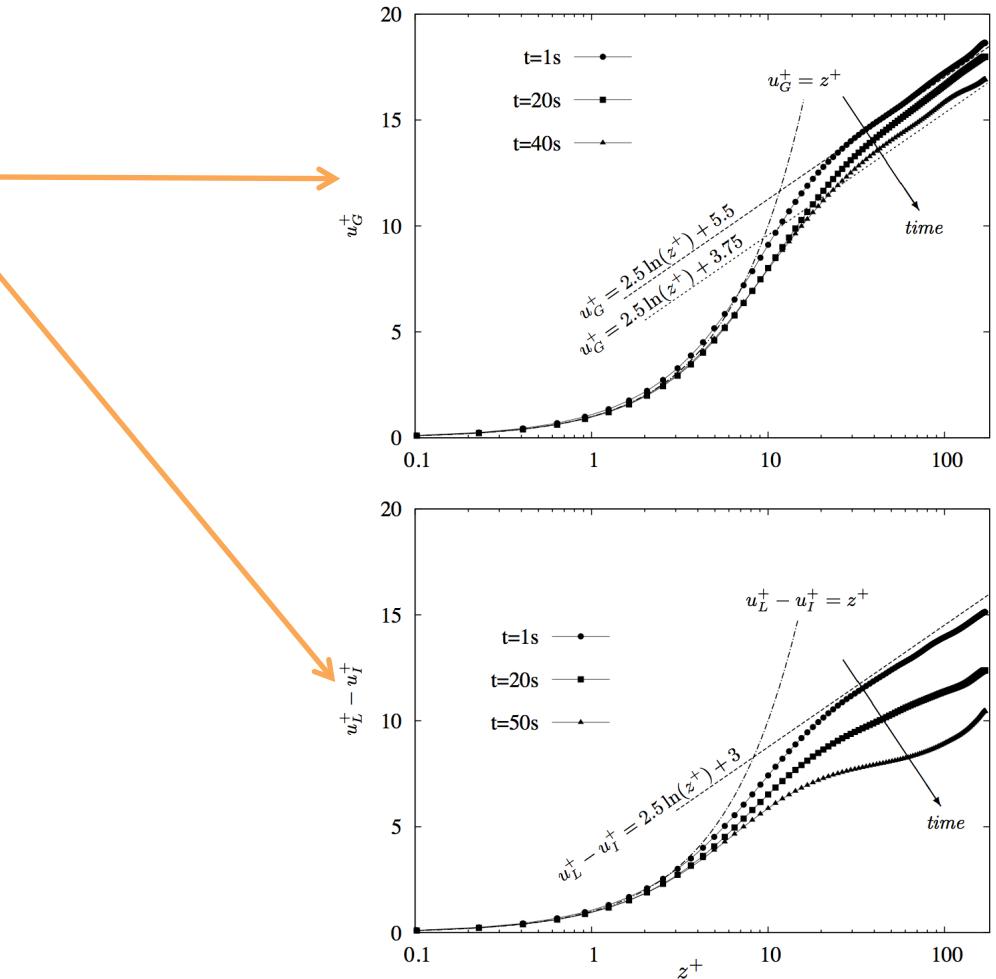
HPC methods for Computational Fluid Dynamics and Astrophysics

CINECA, Bologna, 2-4 Nov. 2016

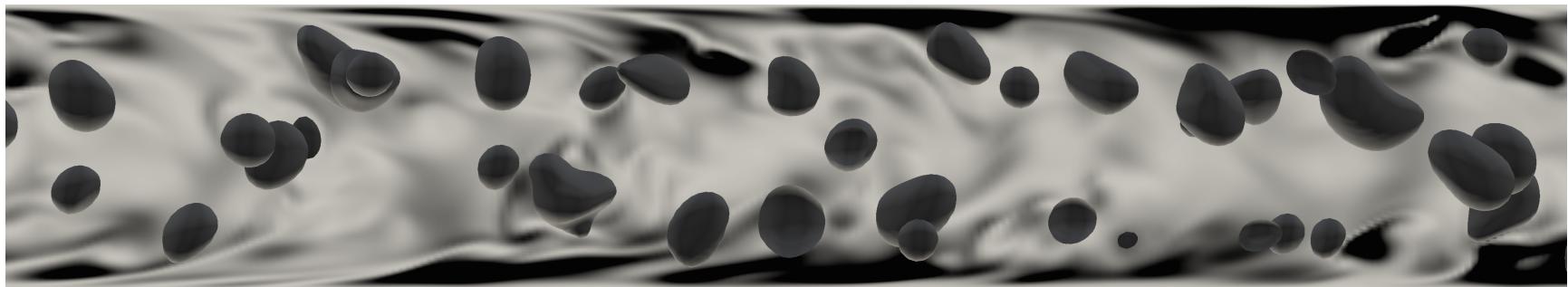
# TURBULENCE MODULATION BY INTERFACE DEFORMATION



As waves develop, we observe a drag increase  
(pressure drag+skin friction)



## PART IV: DROPLETS IN TURBULENCE



MULTIPHASE SYSTEM



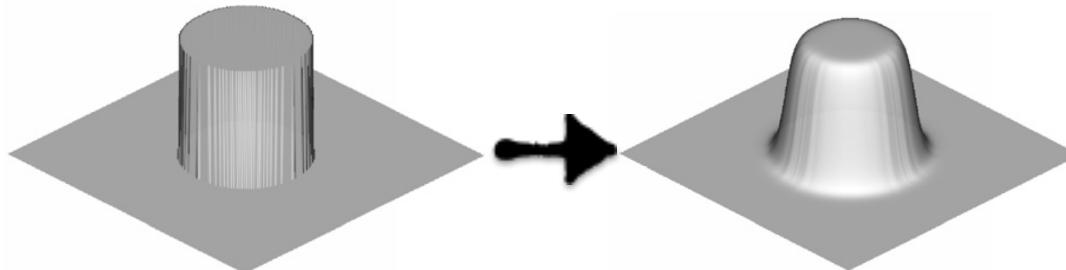
TURBULENCE



MULTIPHASE



Diffuse interface approach (PFM), the sharp interface is replaced by a thin layer of transition:



The Phase Field is described by a transport scalar equation.



Viscosity can depends on  $\Phi$ , using a linear interpolation of viscosity:

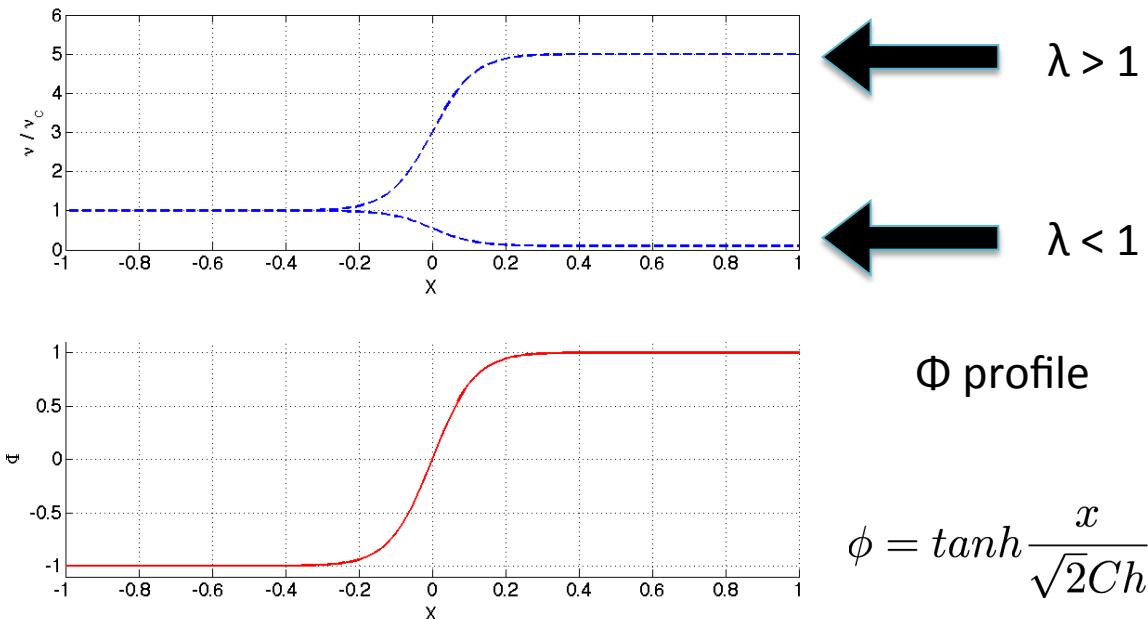
$$\nu(\phi) = \nu_c \frac{1 - \phi}{2} + \nu_d \frac{1 + \phi}{2}$$

Defining  $\lambda$  as:

$$\lambda = \frac{\nu_d}{\nu_c} = \frac{\text{Drop Viscosity}}{\text{Continuos Viscosity}}$$

Viscosity can be rewritten:

$$\nu(\phi) = \nu_c + \nu_c(\lambda - 1) \frac{(\phi + 1)}{2}$$



$$\phi = \tanh \frac{x}{\sqrt{2} Ch}$$

Inserting this expression in the NS equation, the diffusive term can be split as follows:

$$\nabla \cdot [\nu(\phi)(\nabla \mathbf{u} + \nabla \mathbf{u}^T)] = \nu_c \nabla^2 \mathbf{u} + \nu_c \nabla \cdot [(\lambda - 1) \frac{(\phi + 1)}{2} (\nabla \mathbf{u} + \nabla \mathbf{u}^T)]$$

Linear Part

Non Linear Part

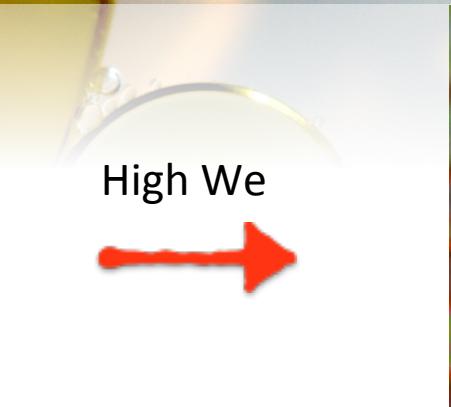
# PHYSICAL PARAMETERS

Weber Number (We):

$$We = \frac{\rho u_\tau^2 H}{\sigma} = \frac{\text{Inertial Force}}{\text{Surface Tension Force}}$$



Low We



High We

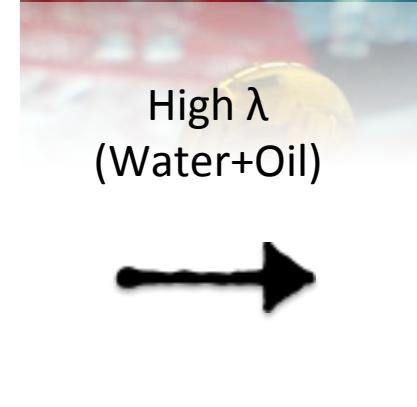


Viscosity Ratio ( $\lambda$ ):

$$\lambda = \frac{\mu_d}{\mu_c} = \frac{\text{Droplet Viscosity}}{\text{Cont.Visocisty}}$$



Low  $\lambda$   
(Water+Hexane)



High  $\lambda$   
(Water+Oil)



## PARAMETERS

$\mu_d$

$\mu_c$

$$\lambda = \frac{\mu_d}{\mu_c}$$

Fixed Parameters

$Re_\tau^*$	<b>150.0</b>
$Ch$	<b>0.0185</b>
$Pe^{**}$	<b>162.00</b>

Pe and Ch numerical parameters.

$$Ch = \frac{\xi}{H} \quad Pe = \frac{u_\tau H}{\Gamma}$$

Grid: 512 x 256 x 257 (N<sub>x</sub>-N<sub>y</sub>-N<sub>z</sub>)

Size:  $4\pi H \times 2\pi H \times 2H$  (L<sub>x</sub>-L<sub>y</sub>-L<sub>z</sub>)

$**Pe \sim 1/Ch$

\* Based on the viscosity of the continuos phase ( $\Phi=1$ )

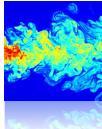
#	We	$\lambda$
S1	0.75	0.01
S2		0.10
S3		1.00
S4		10.0
S5		100.
S6	1.50	0.01
S7		0.10
S8		1.00
S9		10.0
S10		100.
S11	3.00	0.01
S12		0.10
S13		1.00
S14		10.0
S15		100.

**Method:**

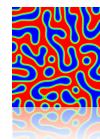
- Direct Numerical Solution (DNS);
- Pseudo-Spectral algorithm;
- N-S: Crank-Nicolson/Adams-Bashforth scheme
- C-H: Crank-Nicolson/Euler scheme

**Boundary Conditions:**

FLOW FIELD



PHASE FIELD



NO SLIP AT THE WALLS

$$\mathbf{u} = 0$$

90° CONTACT ANGLE

$$\frac{\partial \phi}{\partial z} = \frac{\partial^3 \phi}{\partial z^3} = 0$$

PERIODICITY ALONG X and Y

$$\mathbf{u}(0) = \mathbf{u}(L_y)$$

$$\phi(0) = \phi(L_x)$$

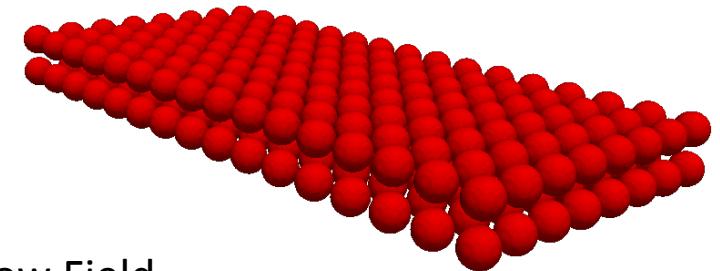
$$\mathbf{u}(0) = \mathbf{u}(L_x)$$

$$\phi(0) = \phi(L_y)$$

**Initial Conditions:**

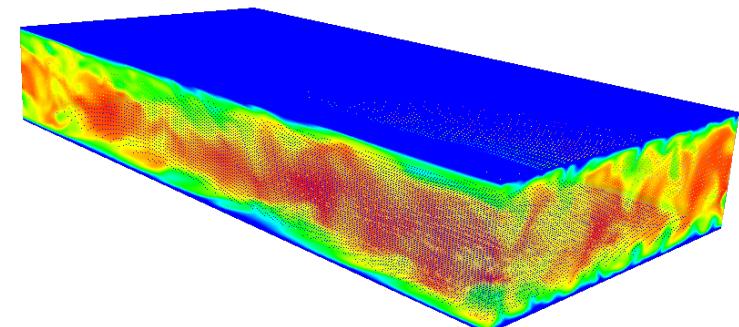
Phase Field

256 Droplets in two arrays

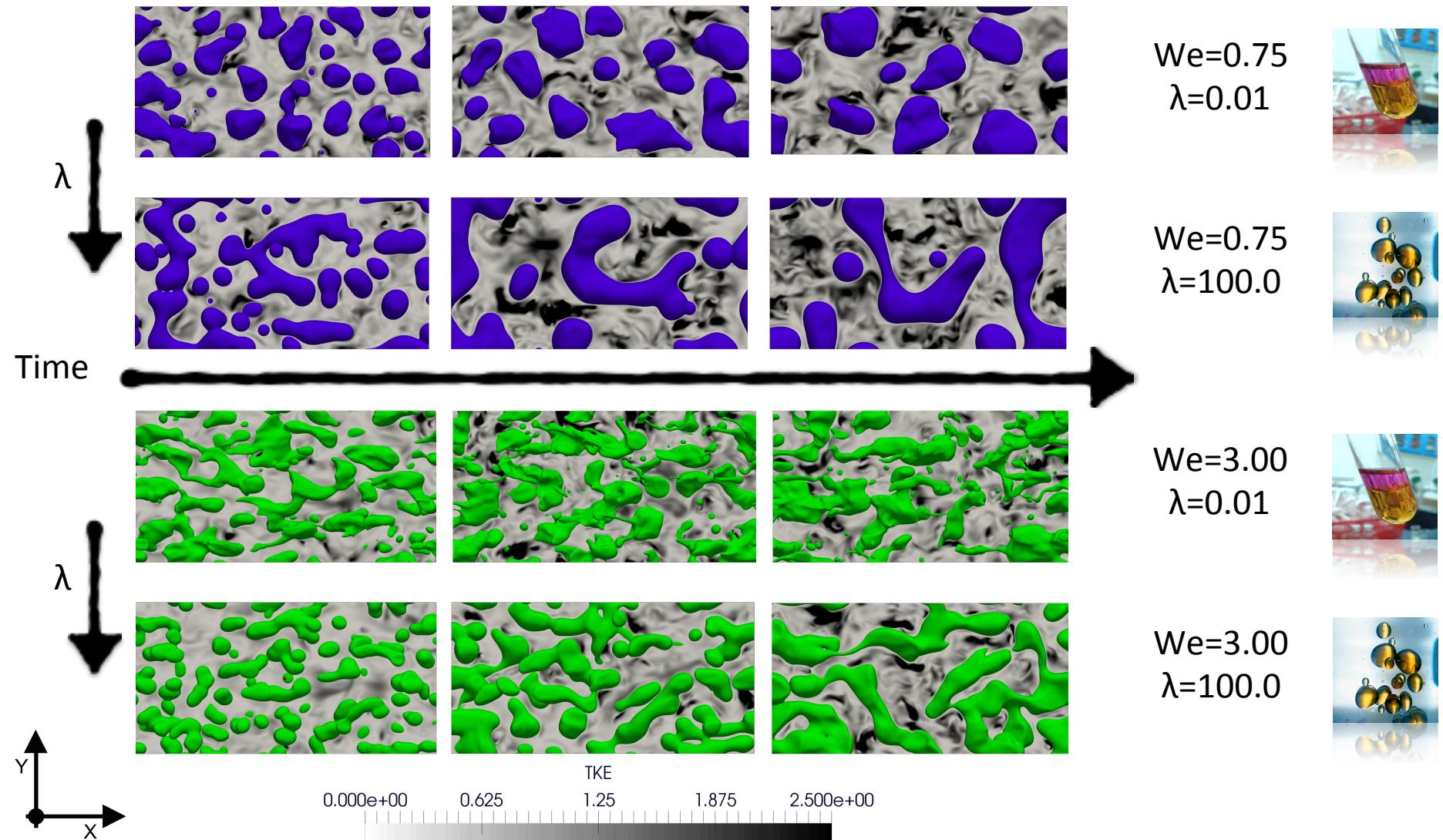
Vol=18.3% d<sup>+</sup>=90 w.u.

Flow Field

-Fully Developed turbulent channel flow  
DNS at Re<sub>τ</sub>=150

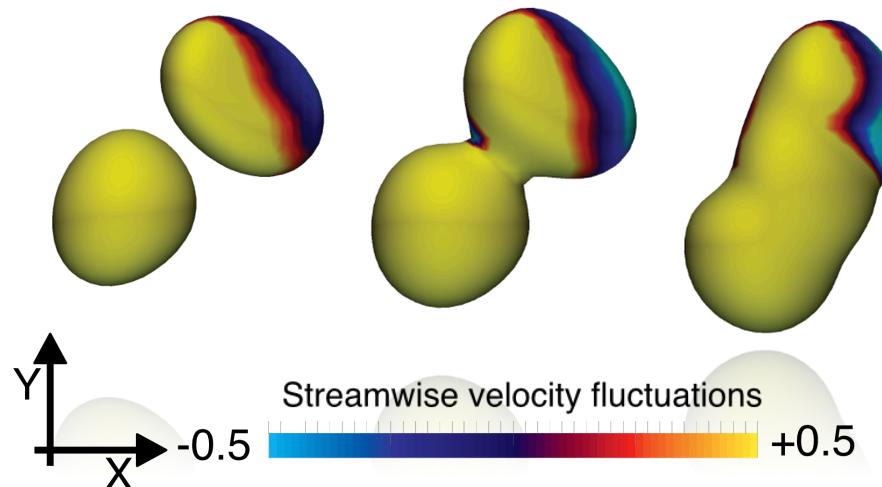


## RESULTS: COLLECTIVE DYNAMICS



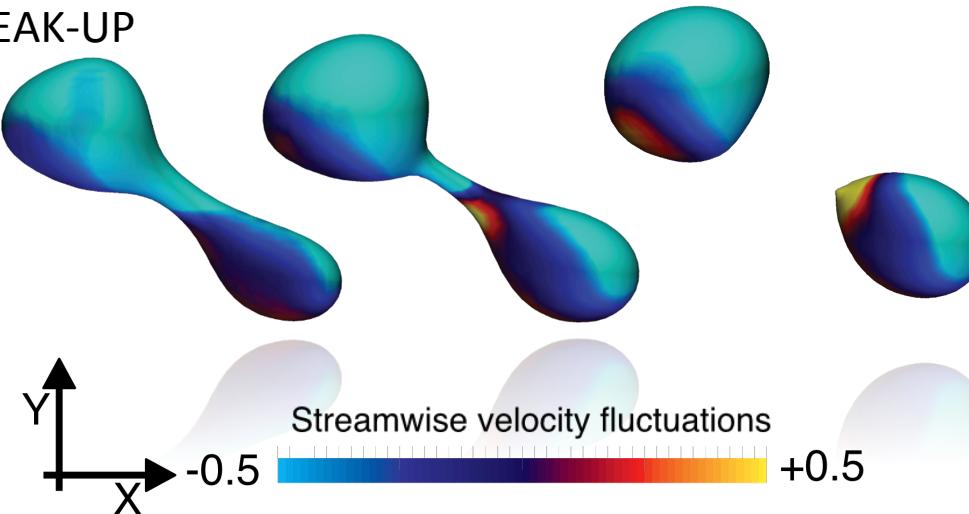
## RESULTS: DROPLETS-DROPLETS INTERACTION

## COALESCENCE



Turbulence fluctuations promote the coalescence phenomena. The new big droplet does not break only if the surface tension is strong enough.

## BREAK-UP

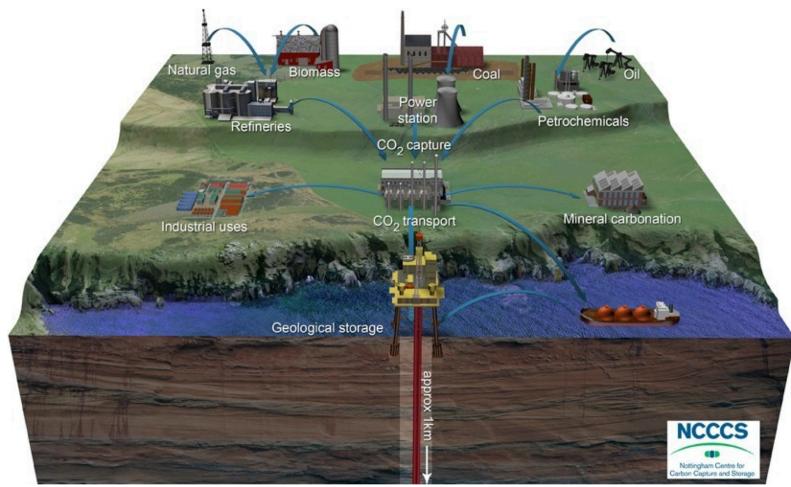


Turbulence fluctuations promote the break-up phenomena; surface tension is not strong enough to keep the single droplet "a single droplet". A bridge is formed and the bridge breaks.

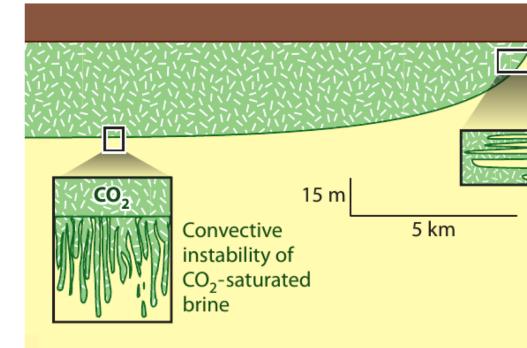
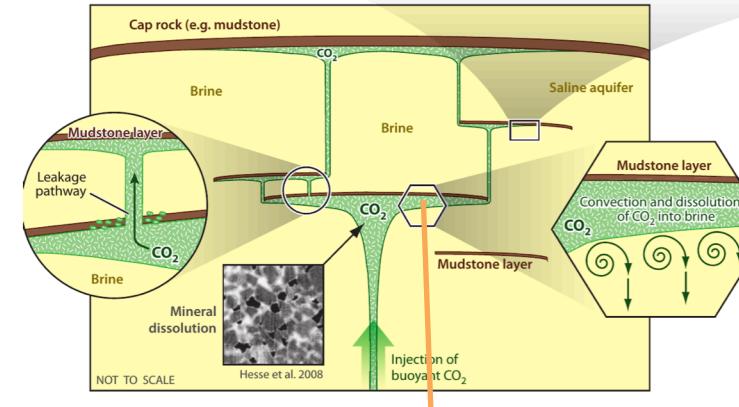
## RESULTS: DROPLETS DYNAMICS



## PART V: GEOLOGICAL CO<sub>2</sub> SEQUESTRATION



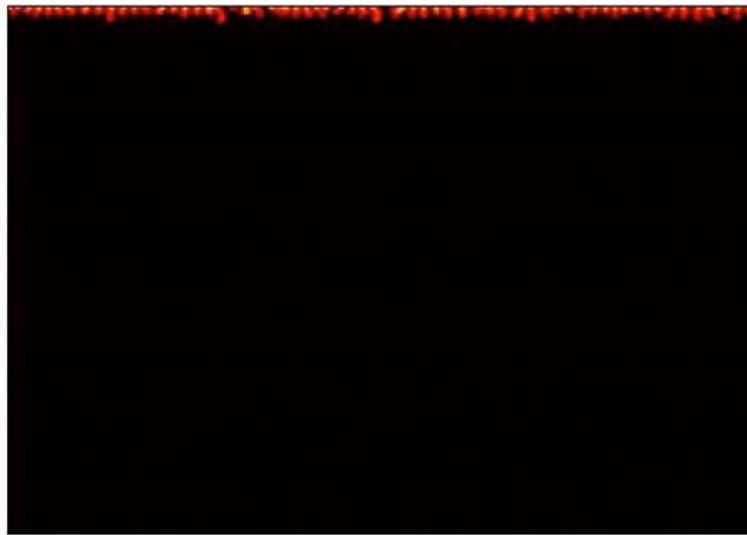
LIQUID CO<sub>2</sub> IS PUMPED BENEATH THE EARTH SURFACE



AT THE BEGINNING, CO<sub>2</sub> MOVES UPWARDS (LOWER DENSITY)  
LATER, IT DISSOLVES INTO BRINE AND MOVES DOWNWARDS

Huppert & Neufeld Annual Rev. Fluid. Mech. 2014

One-sided Config.  
Prescribed Concentration  
@ top wall.  
Zero Flux @ bottom wall



Once this system starts  
filling (i.e. the first finger  
touches the bottom wall)  
Its dynamics becomes  
similar to the two-sided

## Dimensionless equations

$$\frac{\partial u}{\partial x} + \frac{\partial w}{\partial z} = 0$$

$$\mathbf{u} = -\nabla p + \mathbf{k} C$$

$$\frac{\partial C}{\partial t} + \mathbf{u} \cdot \nabla C = \frac{1}{Ra} \left( \gamma \frac{\partial^2 C}{\partial x^2} + \frac{\partial^2 C}{\partial z^2} \right)$$

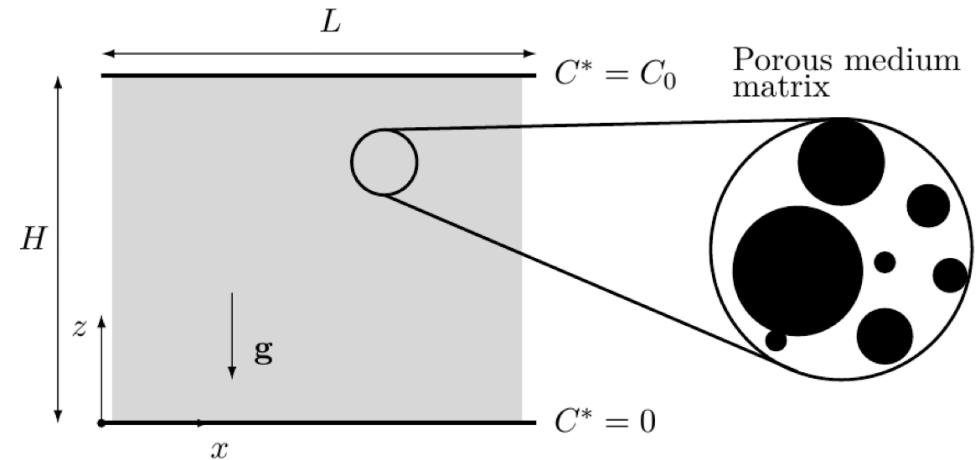
where  $\mathbf{u}, C$  are the fluid velocity and the  $CO_2$  concentration, whereas

$$Ra = \frac{g \Delta \rho K_v H}{\mu \phi D} \text{ and } \gamma = \frac{K_v}{K_h}$$

contain information about fluid and porous medium properties

## Numerical Simulations

We performed 2D Direct Numerical Simulations (DNS) using a pseudo-spectral method (Fourier + Chebyshev) up to **8192x1025** nodes.



## Dimensionless boundary conditions

No-Penetration

$$w = 0 \text{ on } z = 0, 1$$

Fixed concentration

$$C = 0 \text{ on } z = 1 \quad \text{C fixed @ z=1}$$

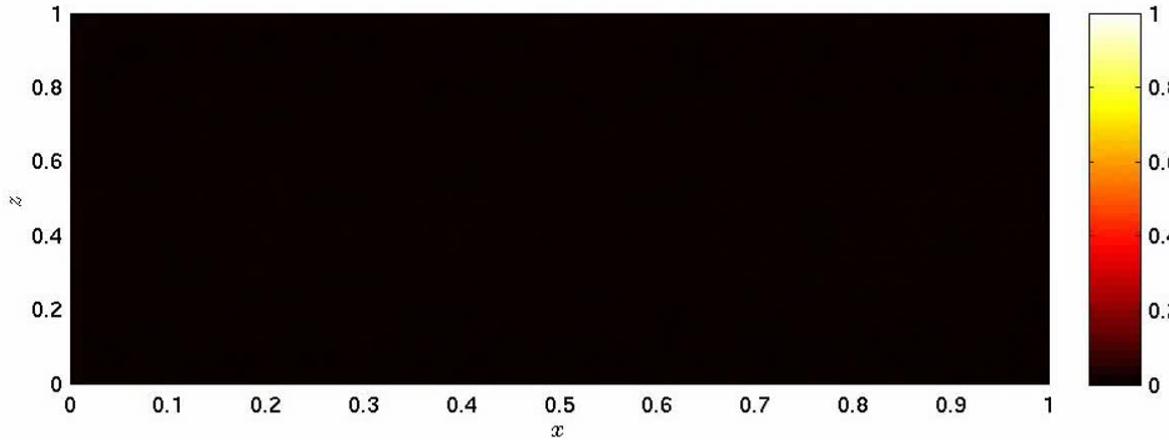
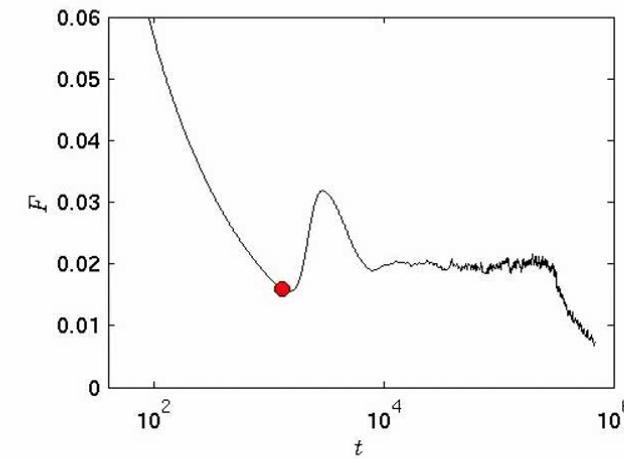
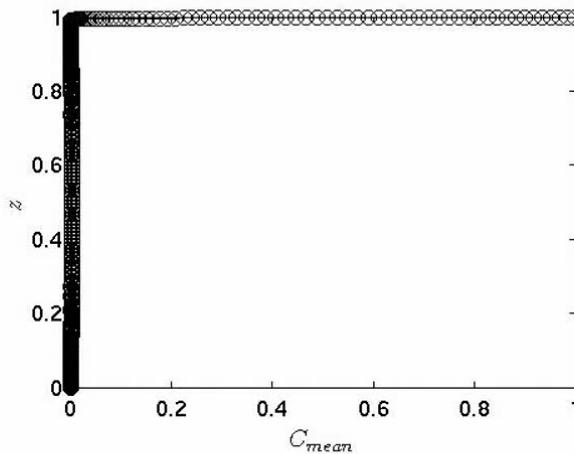
$$C = -1 \text{ on } z = 0 \quad \text{dC/dz fixed @ z=0}$$

Periodic conditions

$$\text{on } x = 0, 1$$

*De Paoli et al., Phys. Fluids (2016)*

# RESULTS: FLUX OF CO<sub>2</sub>



# THANK YOU FOR YOUR ATTENTION